33. FLIGHT PLANNING AND FLIGHT MONIT. -A/C

33.01. FLIGHT PLANS CROSS-COUNTRY FLIGHTS

33.01.01. Navigation plan

33.01.01.01. Selection of routes, speeds, heights

	0	An aircraft is flying at MACH 0.84 at FL 330. The static air temperature is -48°C and
id	57	the headwind component 52 Kt. At 1338 UTC the controller requests the pilot to
		cross the meridian of 030W at 1500 UTC. Given the distance to go is 570 NM, the
		reduced MACH No. should be:
а	0.72	
b	0.78	
С	0.76	
d	0.80	
	1	According to the chart the minimum obstruction clearance altitude (MOCA) is 8500
id	1395	tt. The meteorological data gives an outside air temperature of -20°C at FL 85. The
		when by a met. Station at an elevation of 40001, is 1003 nPa. What is the minimum pressure altitude which should be flown according to the given MOCA?
а	8500	fr
a h	8900	n. f
D C	12800	
ט ה	0200	и. А
<u>a</u>	6200	
id	2 1853	than
а	the he	eighest obstacle.
b	2000	ft above the heighest obstacle within a radius of 600 ft from the aircraft.
С	500 ft	above the heighest obstacle.
d	1000	ft above the heighest obstacle within a radius of 600 m from the aircraft.
	3	How many feet you have to climb to reach FL 75? Given: FL 75; departure
id	1854	aerodrome elevation 1500 ft; QNH = 1023 hPa; temperature = ISA; 1 hPa = 30 ft
а	6300 1	ít.
b	6000	ft.
С	6600	ft.
d	7800	ft.
	4	An aeroplane is flying VFR and approaching position TANGO VORTAC (48°37'N,
id	4393	009°16'E) at FL 055 and magnetic course 090°, distance from VORTAC TANGO
		20 NM. Name the frequency of the TANGO VORTAC.
а	422 k	HZ
b	118.6	0 MHz
С	112.5	0 MHz
d	118.8) MHz

id	5 4578	Flying VFR from VILLINGEN (48°03.5'N, 008°27.0'E) to FREUDENSTADT (48°28.0'N, 008°24.0'E).Determine the minimum altitude within a corridor 5NM left and 5 NM right of the courseline in order to stay 1000 ft clear of obstacles.
а	2900 f	t
b	3900 f	t
С	4200 f	t
d	1500 f	t
id	6 4581	Flying VFR from PEITING (47°48.0'N, 010°55.5'E) to IMMENSTADT (47°33.5'N, 010°13.0'E).Determine the minimum altitude within a corridor 5NM left and 5 NM right of the courseline in order to stay 1000 ft clear of obstacles.
а	6900 f	t
b	5500 f	t
С	6600 f	
d	5300 f	
33	3.01.	01.02. Measurement of tracks and distances
id	7 4576	Flying VFR from VILLINGEN (48°03.5'N, 008°27.0'E) to FREUDENSTADT (48°28.0'N, 008°24.0'E) determine the magnetic course.
а	176°	
b	356°	
С	004°	
d	185°	
id	8 4577	Flying VFR from VILLINGEN (48°03.5'N, 008°27.0'E) to FREUDENSTADT (48°28.0'N, 008°24.0'E) determine the distance.
а	24 NM	
b	46 NM	
С	28 NM	
d	24 km	
id	9 4579	Flying VFR from PEITING (47°48.0'N, 010°55.5'E) to IMMENSTADT (47°33.5'N, 010°13.0'E) determine the magnetic course.
а	063°	
b	243°	
С	257°	
d	077°	
id	10 4580	Flying VFR from PEITING (47°48.0'N, 010°55.5'E) to IMMENSTADT (47°33.5'N, 010°13.0'E) determine the distance.
а	32 NM	
b	46 NM	
С	58 NM	
d	36 NM	
id	11	The average magnetic course from C (62°N020°W) to B (58°N004°E) is
יע פ	100°	
u h	119°	
r c	0900	
с А	118°	
u	110	

•	12	The average true course from C (62°N020°W) to B (58°N004°E) is
id	5738	
а	120°	
b	119°	
C	099°	
a	109°	
hi	13	The initial magnetic course from C (62°N020°W) to B (58°N004°E) is
a	116°	
b	080°	
с	098°	
d	113°	
	14	The initial true course from C (62°N020°W) to B (58°N004°E) is
id	5740	
а	098°	
b	116°	
С	080°	
d	278°	
	15	The distance (NM) from A (64°N006°E) to C (62°N020°W) is
id	5741	
a ⊾	7 20	
a	1500	
d d	1290	
	1410	The second second (is second a (240) 10000 () (s. 0 (000) 10000) () is
id	1 6 5742	The average magnetic course from A (64°N006°E) to C (62°N020°W) is
а	271°	
b	259°	
С	247°	
d	279°	
	17	The average true course from A (64°N006°E) to C (62°N020°W) is
id	5743	
а	271°	
b	247°	
С	259°	
d	079°	
	18	The initial magnetic course from A (64°N006°E) to C (62°N020°W) is
id O	5/44	
a h	202 267°	
с С	271°	
ч Ч	275°	
u	215	

:-1	19	The initial true course from A (64°N006°E) to C (62°N020°W) is
ıa a	5745 271°	
a h	275°	
C C	273 267°	
о Ь	207 246°	
	2-10	The distance (NNA) from $O_{1}(200N(0.000)M)$ to $D_{1}(200N(0.040E))$ is
id	20 5746	The distance (NW) from C ($62^{\circ}N020^{\circ}W$) to B ($58^{\circ}N004^{\circ}E$) is
а	775	
b	725	
С	700	
d	760	
3:	3.01.	01.04. Comp. of headings, ground speeds
_	21	On a given path, it is possible to chose between four flight levels (FL), each
id	58	associated with a mandatory flight Mach Number (M). The flight conditions, static
		air temperature (SAT) and headwind component (HWC) are given below: FL 370 -
		$M = 0.80 \text{ Is} = -60^{\circ}\text{C}$ HWC = -15 kt FL 330 - M = 0.78 Is = -60°C HWC = -5 kt FL 290 - M = 0.80 Ts = -55°C HWC
а	FI 270	
b	FL290	
c	FL330	
d	FL370	
	22	A twin-jet aeroplane carries out the WASHINGTON-PARIS flight When it reaches
id	5 9	point K (35°N - 048°W) a non-mechanical event makes the Captain consider
		rerouting to one of the three following fields. The flight conditions are: - from K to
		BERMUDAS (distance 847NM, headwind component=18 kt) - from K to SANTA
~	DEDN	MARIA (distance 1112 NM, taliwind component=120 kt) - nom k to GANDER
d h		
D C	BERM	
ч	Eithor	
	23	An aeroplane flies at an airspeed of 380 kt. It flies from A to B and back to A
id	23 70	Distance $AB = 480$ NM. When going from A to B, it experiences a headwind
		component = 60 kt. The wind remains constant. The duration of the flight will be:
а	3h 00i	nin
b	2h 35	min
С	2h 10	nin
d	2h 32ı	nin
hi	24	Given : true track 017; W/V 340/30; TAS 420 kt Find : wind correction angle (WCA)
.u		
a h		$-2^{\circ} \cdot GS 396 \text{ kt}$
2		-2° · GS 426 kt
ч С	WCA	+2° · GS 416 kt
u		12,00110 M

id	25 2175	Flight planning chart for an aeroplane states, that the time to reach the cruising level at a given gross mass is 36 minutes and the distance travelled is 157 NM (zero-wind). What will be the distance travelled with an average tailwind component of 60kt ?
а	193 N	M
b	128 N	M
С	: 157 N	M
d	228 N	M
id	26 2181	You are flying a constant compass heading of 252°. Variation is 22°E, deviation is 3°W and your INS is showing a drift of 9° right. True track is ?
а	242°	
b	224°	
С	280°	
d	262°	
id	27 4385	Given: True course (TC) 017°, W/V 340°/30 kt, True air speed (TAS) 420 kt Find: Wind correction angle (WCA) and ground speed (GS)
а	WCA	-2°, GS 426 kt
b	WCA	+2°, GS 396 kt
С	WCA	-2°, GS 396 kt
d	I WCA	+2°, GS 416 kt
3	3.01	.01.05. Completion of pre-flight portion
id	28 5540	An executive pilot is to carry out a flight to a French aerodrome, spend the night there and return the next day. Where will he find the information concerning parking

and landing fees ?

a in the FAL section of the French Aeronautical Information Publication (AIP)

- ${\bf b}\,$ in the AGA chapter of the French Aeronautical Information Publication (AIP)
- ${\bf c}\,$ in the GEN chapter of the French Aeronautical Information Publication (AIP)
- **d** by telephoning the aerodrome's local chamber of commerce, this type of information not being published

33.01.02. Fuel plan

33.01.02.01. Computation of planned fuel usage

- (For this question use Flight Planning Manual MEP1 Figure 3.1) A flight is to be made from one airport (elevation 3000 ft) to another in a multi engine piston aireroplane (MEP1). The cruising level will be FL 110. The temperature at FL 110 is ISA 10° C. The temperature at the departure aerodrome is -1° C. Calculate the fuel to climb with mixture rich.
- a 9 US gallon
- b 6 US gallon
- c 12 US gallon
- d 3 US gallon

30 (For this question use Flight Planning Manual MEP1 Figure 3.6) A flight is to be

- id 2041 made to an airport, pressure altitude 3000 ft, in a multi engine piston aireroplane (MEP1). The forecast OAT for the airport is -1° C. The cruising level will be FL 110, where OAT is -10° C. Calculate the still air descent distance for: 145 KIAS Rate of descent 1000 ft/min Gears and flaps up
- **a** 25 NM
- **b** 29 NM
- **c** 36 NM
- d 20 NM
- **31** (For this Question use Flight Planning & Monitoring SEP1 Fig. 2.2) Given: FL 75 oAT +10°C Lean mixture 2300 RPM Find: Fuel flow in gallons per hour (GPH) and TAS.
- a 71.1 GPH TAS: 143 kt
- b 11.6 GPH TAS: 143 kt
- c 11.6 GPH TAS: 160 kt
- d 68.5 GPH TAS: 160 kt

32 (For this Question use Flight Planning & Monitoring SEP1 Fig. 2.1)Given: FL 75 ¹²⁵⁶⁶ OAT: +5°C During climb: average head wind component 20 kt Take-off from MSL with the initial mass of 3 650 lbs. Find: Time and fuel to climb.

- a 7 min. 2,6 USG
- **b** 10 min. 3,6 USG
- c 9 min. 3,3 USG
- **d** 9 min. 2,7 USG

33 (For this Question use Flight Planning & Monitoring SEP1 Fig. 2.1) Given: FL 75
 id 2567 OAT: +5°C During climb: average head wind component 20 kt Take-off from MSL with the initial mass of 3 650 lbs. Find: Still air distance (NAM) and ground distance (NM) using the graph "time, fuel, distance to climb".

a 18 NAM. 15 NM.

- **b** 16 NAM. 18 NM.
- c 18 NAM. 13 NM.
- d 14 NAM. 18 NM.

34 (For this Question use Flight Planning & Monitoring SEP 1, Fig. 2.1) Given: Take-^{id} ⁴³⁹¹ off mass 3500 lbs, departure aerodrome pressure altitude 2500 ft, OAT +10°C, First cruising level: FL 140, OAT -5°C Find the time, fuel and still air distance to climb.

a 22 min, 6.7 GAL, 45 NAM

- **b** 24 min, 7.7 GAL, 47 NAM
- c 16.5 min, 4.9 GAL, 34.5 NAM
- **d** 23 min, 7.7 GAL, 50 NAM

35 (For this Question use Flight Planning & Monitoring SEP1, Fig. 2.4) Given:

- ^{id} 4392 Aeroplane mass at start-up 3663 lbs, Aviation gasoline (density 6 lbs/gal)-fuel load
 74 gal, Take-off altitude sea level, Headwind 40 kt, Cruising altitude 8000 ft, Power setting full throttle 2300 RPM 20°C lean of peak EGT Calculate the range.
- **a** 547.5 NM
- **b** 844 NM
- **c** 730 NM
- d 633 NM

	36	The fuel burn off is 200 kg/h with a relative fuel density of 0,8. If the relative density
Ia	5517	IS 0,75, the fuel burn will be:
a	213 k	g/h
b	200 K	g/n
C	188 K	g/h
a	267 K	
id	37 5533	In the cruise at FL 155 at 260 kt TAS, the pilot plans for a 500 feet/min descent in order to fly overhead MAN VOR at 2 000 feet (QNH 1030). TAS will remain constant during descent, wind is negligible, temperature is standard. The pilot must start the descent at a distance from MAN of:
а	140 N	Μ
b	120 N	M
С	110 N	M
d	130 N	Μ
id	38 5539	An aircraft is in cruising flight at FL 095, IAS 155kt. The pilot intends to descend at 500 ft/min to arrive overhead the MAN VOR at 2 000 FT (QNH 1 030hPa). The TAS remains constant in the descent, wind is negligeable, temperature standard. At which distance from MAN should the pilot commence the descent?
а	48 NN	1
b	42 NN	1
С	40 NN	1
d	45 NN	1
id	39 5674	(For this Question use Fuel Planning MRJT1) Given : Distance C - D : 3200 NM Long Range Cruise at FL 340 Temperature Deviation from ISA : +12°C Tailwind component : 50 kt Gross mass at C : 55 000 kg The fuel required from C - D is :
а	17 50	0 kg
b	14 20	0 kg
С	17 80	0 kg
d	14 50	0 kg
id	40 5675	(For this Question use Fuel Planning MRJT1) Given : Distance C - D : 680NM Long Range Cruise at FL340 Temperature Deviation from ISA : 0° C Headwind component : 60 kt Gross mass at C : 44 700 kg The fuel required from C - D is :
а	3400	kg
b	3700	kg
С	3100	kg
d	4000	kg
id	41 5676	(For this Question use Fuel Planning MRJT1) Given : Brake release mass : 58 000 kg Temperature : ISA + 15 The fuel required to climb from an aerodrome at elevation 4000 ft to FL300 is :
а	1350	kg
b	1400	kg
С	1450	kg
d	1250	kg

id	42 5677	(For this Question use Fuel Planning MRJT1) Given : Brake release mass : 62 000 kg Temperature : ISA + 15°C The fuel required for a climb from Sea Level to
		FL330 is :
а	1800	kg
b	1650	kg
С	1750	kg
d	1700	kg
	43	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 1200 NM
Id	5679	Cruise Mach 0.78 at FL300 Temperature Deviation from ISA : -14°C Tailwind
		:
а	5850	ka
h	6150	ka
~ c	7300	ka
о Ь	7050	ka
	<u> </u>	(For this Question use Fuel Planning MR IT1) Given : Distance C - D : 540 NM
id	 5680	Cruise 300 KIAS at FL 210 Temperature Deviation from ISA : +20°C Headwind
		component: 50 kt Gross mass at C: 60 000 kg The fuel required from C to D
		is :
а	3680	kg
b	4620	kg
С	3350	kg
d	4242	kg
	15	
	4 J	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM
id	40 5681	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is :
id a	5681 1940	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is :
^{id} a b	5681 1940	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg
id a b c	5681 1940 1810	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg
id a b c d	5681 1940 1810 2800 2670	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg
id a b c d	5681 1940 1810 2800 2670	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical
id a b c d id	5681 1940 1810 2800 2670 46 5711	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA +
id a b c d id	5681 1940 1810 2800 2670 46 5711	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and
id a b c d id	5681 1940 1810 2800 2670 46 5711	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are :
id a b c d id	5681 1940 1810 2800 2670 46 5711 (a) 20	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 7hr 00 min
id a b c d id a b	 5681 1940 1810 2800 2670 46 5711 (a) 20 (a) 16 	 (For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 7hr 00 min 200 kg (b) 6 hr 20 min
id a b c d id a b c	5681 1940 1810 2800 2670 46 5711 (a) 2((a) 16 (a) 17	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 7hr 00 min 5200 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 10 min
id abcd id abcd	5681 1940 1810 2800 2670 46 5711 (a) 20 (a) 10 (a) 17 (a) 17 (a) 17	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 2000 kg (b) 7hr 00 min 5200 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 10 min 7600 kg (b) 6 hr 50 min
id abcd id abcd	 5681 1940 2800 2670 46 5711 (a) 20 (a) 16 (a) 17 	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : -40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 7hr 00 min 5200 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 20 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical with a the following apply is the divide severe and 20 km T
id a b c d id a b c d id	 5681 1940 1810 2800 2670 46 5711 (a) 20 (a) 10 (a) 17 (a) 17 (a) 17 (a) 17 5712 	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 20 min 3200 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 20 kt Temperature: ISA + 15°C Brake release mass: 64700 kg The (a) trip fuel and (b) trip time respectively
id abcd id abcd id	 5681 1940 2800 2670 46 5711 (a) 20 (a) 10 (a) 17 (a) 17 (a) 17 5712 	<pre>(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 10 min 7000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 20 kt Temperature: ISA + 15°C Brake release mass: 64700 kg The (a) trip fuel, and (b) trip time respectively are :</pre>
id abcdid abcdid a	5681 1940 2800 2670 46 5711 (a) 26 (a) 16 (a) 17 5712 (a) 16	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : -40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 20 kt Temperature: ISA + 15°C Brake release mass: 64700 kg The (a) trip fuel, and (b) trip time respectively are : 3200 kg (b) 6 hr 20 min
id abcdid abcdid ab	 5681 1940 1810 2800 2670 46 5711 (a) 20 (a) 10 (a) 17 (a) 17 (a) 17 (a) 16 (a) 16 (a) 16 (a) 16 (a) 16 (a) 16 	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 20 kt Temperature: ISA + 15°C Brake release mass: 64700 kg The (a) trip fuel, and (b) trip time respectively are : 5200 kg (b) 6 hr 20 min 5200 kg (b) 6 hr 20 min 5200 kg (b) 6 hr 20 min 5200 kg (b) 6 hr 20 min
id abcdid abcdid abc	5681 1940 2800 2670 46 5711 (a) 20 (a) 17 (a) 17 (a) 17 (a) 16	(For this Question use Fuel Planning MRJT1) Given : Distance B - C : 350 NM Cruise 300 KIAS at FL 210 Temperature : - 40°C Tailwind component : 70 kt Gross mass at B : 53 200 kg The fuel required from B - C is : kg kg kg kg (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 15 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 20 min 7000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 20 kt Temperature: ISA + 15°C Cruise altitude: 35000 ft Landing mass: 50000 kg The (a) trip fuel and (b) trip time respectively are : 0000 kg (b) 6 hr 50 min (For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Head wind component: 20 kt Temperature: ISA + 15°C Brake release mass: 64700 kg The (a) trip fuel, and (b) trip time respectively are : 2000 kg (b) 6hr 20 min 3000 kg (b) 6hr 20 min 3000 kg (b) 6hr 45 min

id	48 5713	(For this Question use Fuel Planning MRJT1) For a flight of 1900 ground nautical miles the following apply : Head wind component 10 kt Temperature ISA -5°C Trip fuel available 15000 kg Landing mass 50000kg What is the minimum cruise level (pressure altitude) which may be planned ?
а	17000	ft
b	22000	ft
С	14000	ft
d	10000	ft
id	49 5714	(For this Question use Fuel Planning MRJT1) Given the following : Head wind component 50 kt Temperature ISA + 10°C Brake release mass 65000kg Trip fuel available 18000kg What is the maximum possible trip distance ?
а	3480 1	NM
b	3100 1	NM
С	2740 N	IM
d	2540 1	IM
id	50 5718	(For this Question use Fuel Planning MRJT1) For a flight of 2800 ground nautical miles the following apply : Tail wind component 45kt Temperature ISA - 10°C Cruise altitude 29000ft Landing mass 55000kg The (a) trip fuel (b) trip time respectively are :
а	(a) 20	000kg (b) 6hr 40 min
b	(a) 18	000kg (b)5hr 50 min
С	(a) 17	100kg (b) 6hr 07 min
d	(a) 16	000kg (b) 6hr 25 min
id	51 5720	(For this Question use Fuel Planning MRJT1) The following apply: Temperature ISA +15°C Brake release mass 62000kg Trip time 5hr 20 min What is the trip fuel ?
а	13800	kg
b	13000	kg
с	13200	kg
d	13500	kg
id	52 5721	(For this Question use Fuel Planning MRJT1) For a flight of 2400 ground nautical miles the following apply : Temperature ISA -10°C Cruise altitude 29000ft Landing mass 45000kg Trip fuel available 16000kg What is the maximum headwind component which may be accepted ?
а	0	
b	15kt	
С	70kt	
d	35 kt	
id	53 5722	(For this Question use Fuel Planning MRJT1) The following apply : Tail wind component 10kt Temperature ISA +10°C Brake release mass 63000kg Trip fuel available 20000kg What is the maximum possible trip distance ?
а	3500 N	M
b	3640 1	M
C	3740 N	M
d	3250 N	JM

- 54(For this Question use Fuel Planning MRJT1) For a flight of 2400 ground nautical
miles the following apply : Tail wind component 25 kt Temperature ISA -10°C
Cruise altitude 31000ft Landing mass 52000kg The (a) trip fuel and (b) trip
time respectively are :
- a (a) 14200kg (b) 5 hr 30 min
- **b** (a) 16200kg (b) 5 hr 45 min
- c (a) 13600kg (b) 6 hr 30 min
- **d** (a) 12000kg (b) 5 hr 15 min

33.01.02.02. Fuel for holding or diversion

- **55** Given: Dry operating mass (DOM)= 33510 kg Load= 7600 kg Final reserve fuel= ^{id} ¹⁸⁵⁶ 983 kg Alternate fuel= 1100 kg Contingency fuel 102 kg The estimated landing mass at alternate should be :
- **a** 42312 kg.
- **b** 42093 kg.
- **c** 42210 kg.
- d 42195 kg.
- **56** Given: Dry operating mass (DOM)= 33000 kg Load= 8110 kg Final reserve fuel= **id** 1857 Given: Dry operating mass (DOM)= 33000 kg Load= 8110 kg Final reserve fuel= **983 kg Alternate fuel= 1100 kg Contingency fuel 102 kg The estimated landing** mass at alternate should be :
- **a** 42312 kg.
- **b** 41110 kg.
- **c** 42210 kg.
- d 42195 kg.
- **57** Given: Dry operating mass (DOM)= 33510 kg Load= 7600 kg Trip fuel (TF)= 2040 kg Final reserve fuel= 983 kg Alternate fuel= 1100 kg Contingency fuel= 5% of trip fuel Which of the listed estimated masses is correct?
- a Estimated take-off mass= 45233 kg.
- b Estimated landing mass at destination= 43295 kg.
- c Estimated landing mass at destination= 43193 kg.
- d Estimated take-off mass= 43295 kg.
- **58** (For this Question use Fuel Planning MRJT1) HOLDING PLANNING The fuel required for 30 minutes holding, in a racetrack pattern, at PA 1500 ft, mean gross mass 45 000 kg, is :
- **a** 1010 kg
- b 1090 kg
- **c** 1310 kg
- d 2180 kg

59 (For this Question use Fuel Planning MRJT1) HOLDING PLANNING The fuel required for 45 minutes holding, in a racetrack pattern, at PA 5000 ft, mean gross mass 47 000 kg, is :

- a 1635 kg
- **b** 1090 kg
- **c** 1690 kg
- **d** 1125 kg

60 (For this Question use Fuel Planning MRJT1) Given: Distance to Alternate 450 NM id 5684 Landing mass at Alternate : 45 000 kg Tailwind component : 50 kt The Alternate fuel required is :

- **a** 2900 kg
- **b** 2750 kg
- **c** 3050 kg

d 2500 kg

61 (For this Question use Fuel Planning MRJT1) Given : Distance to Alternate : 400 ^{id} 5685 NM Landing mass at Alternate : 50 000kg Headwind component : 25 kt The alternate fuel required is :

- **a** 2650 kg
- **b** 2550 kg
- **c** 2900 kg

d 2800 kg

33.01.02.03. Reserves

62 A public transport aeroplane with reciprocating engines, is flying from PARIS to LYON. The final reserve corresponds to:

a 45 minutes at holding speed

b 2 hours at cruise consumption

c 1 hour at holding speed

d 30 minutes at holding speed

- 63 In a flight plan when the destination aerodrome is A and the alternate aerodrome is ^{id} ⁶¹ B, the final reserve fuel for a turbojet engine aeroplane corresponds to:
- a 15 minutes holding 2,000 feet above aerodrome A
- b 30 minutes holding 2,000 feet above aerodrome B

c 30 minutes holding 1,500 feet above aerodrome B

d 30 minutes holding 1,500 feel above aerodrome A

- 64 Following in-flight depressurisation, a turbine powered aeroplane is forced to divert
- id 1199 to an en-route alternate airfield. If actual flight conditions are as forecast, the minimum quantity of fuel remaining on arrival at the airfield will be:
- a at least equivalent to 45 minutes flying time
- ${\bf b}\,$ at least equivalent to the quantity required to fly to another aerodrome in the event that weather conditions so require
- c laid down by the operator, with the quantity being specified in the operating manual

d at least equivalent to 30 minutes flying time

33.01.02.04. Total fuel requirements for flight

- 65 The Trip Fuel for a jet aeroplane to fly from the departure aerodrome to the destination aerodrome is 5 350 kg. Fuel consumption in holding mode is 6 000 kg/h. The quantity of fuel which is needed to carry out one go-around and land on the alternate airfield is 4 380 kg. The destination aerodrome has a single runway. What is the minimum quantity of fuel which should be on board
- **a** 13 000 kg
- b 13 230 kg
- **c** 12 700 kg
- **d** 10 000 kg

id	60 63	For turbojet engine driven aeroplane, given: Taxi tuel 600 kg Fuel flow for cruise 10,000 kg/b Euel flow for holding 8,000 kg/b Alternate fuel 10,200
		kg Planned flight time to destination 6 h Forecast visibility at destination 2000 m
		The m
а	77 80	0 kg
b	76 10	0 kg
С	80 50	0 kg
d	79 20	0 kg
:	67	(For this Question use Flight Planning & Monitoring MEP1) A flight has to be made
Ia	2539	the taxi, and an additional 13 minutes at cruise condition to account for climb and
		descent. Calculated time from overhead to overhead is 1h47min. Powersetting is
		45%, 2600 RPM. Calculated reserve fuel is 30% of the tri
а	37 US	S gallons.
b	47 US	S gallons.
С	60 US	S gallons.
d	470 L	IS gallons.
	68	(For this Question use Flight Planning & Monitoring MEP1) A flight has to be made
id	2540	with a multi engine piston aeroplane. For the fuel calculations take 5 US gallons for
		descent. Calculated time overhead to overhead is 2h37min. Powersetting is 65%.
		2500 RPM. Calculated reserve fuel is 30% of the trip fue
а	91 US	S gallons.
h		
D	86 US	S gallons.
с С	86 US 76 US	S gallons. S gallons.
c d	86 US 76 US 118 U	S gallons. S gallons. IS gallons.
c c d	86 US 76 US 118 U 69	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following
c d id	86 US 76 US 118 U 69 5541	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg
d d id	86 US 76 US 118 U 69 5541	S gallons. S gallons. JS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg
d id b	86 Us 76 Us 118 U 69 5541 1000	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg
d d id a b c	86 US 76 US 118 U 69 5541 1000 800 k 700 k	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg
d d id a b c d	86 US 76 US 118 U 69 5541 1000 800 k 700 k 500 k	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g
d d id a b c d 3	86 US 76 US 118 U 69 5541 1000 800 k 700 k 500 k 3.01 .	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log
d d id a b c d 3	86 US 76 US 118 U 69 5541 1000 800 k 500 k 500 k 3.01.	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log Given maximum allowable take-off mass 64 400 kg maximum landing mass 56
d id a b c d 3 id	86 US 76 US 118 L 69 5541 1000 800 k 700 k 500 k 3.01. 70 2043	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 64 400 kg maximum landing mass 56 200 kg maximum zero fuel mass 53 000 kg dry operating mass 35 500 kg
d id b c d 3 id	86 US 76 US 118 U 69 5541 1000 800 k 500 k 500 k 3.01. 70 2043	S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 64 400 kg maximum landing mass 56 200 kg maximum zero fuel mass 53 000 kg dry operating mass 35 500 kg estimated load 14 500 kg estimated trip fuel 4 900kg minimum take-off fuel 7 400
d id a b c d 3 id id	86 US 76 US 118 U 69 5541 1000 800 k 500 k 500 k 3.01. 70 2043	 S gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g 02.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 53 000 kg dry operating mass 35 500 kg estimated load 14 500 kg estimated trip fuel 4 900kg minimum take-off fuel 7 400 kg Find the maximum allowable take-off fuel:
d id a b c d 3 id id a	86 US 76 US 118 U 69 5541 1000 800 k 500 k 3.01. 70 2043 8 6000	s gallons. S gallons. IS gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 64 400 kg maximum landing mass 56 200 kg maximum zero fuel mass 53 000 kg dry operating mass 35 500 kg estimated load 14 500 kg estimated trip fuel 4 900kg minimum take-off fuel 7 400 kg Find the maximum allowable take-off fuel: kg
d id a b c d 3 id a b	86 US 76 US 118 U 69 5541 1000 800 k 500 k 500 k 3.01. 70 2043 8 600 11 40	s gallons. S gallons. S gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 64 400 kg maximum landing mass 56 200 kg maximum zero fuel mass 53 000 kg dry operating mass 35 500 kg estimated load 14 500 kg estimated trip fuel 4 900kg minimum take-off fuel 7 400 kg Find the maximum allowable take-off fuel: kg 0 kg
d id a b c d 3 id id a b c .	86 US 76 US 118 U 69 5541 1000 800 k 500 k 500 k 500 k 3.01. 70 2043 8 600 11 40 14 40	s gallons. S gallons. S gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g 02.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 64 400 kg maximum landing mass 56 200 kg maximum zero fuel mass 53 000 kg dry operating mass 35 500 kg estimated load 14 500 kg estimated trip fuel 4 900kg minimum take-off fuel 7 400 kg Find the maximum allowable take-off fuel: kg 0 kg 0 kg
d id a b c d 3 id id a b c d d c d d c d	86 US 76 US 118 U 69 5541 1000 800 k 500 k 500 k 3.01. 70 2043 8 600 11 40 14 40 14 10	s gallons. S gallons. S gallons. You are to determine the maximum fuel load which can be carried in the following conditions : - dry operating mass : 2800 kg - trip fuel : 300 kg - payload : 400 kg - maximum take-off mass : 4200 kg - maximum landing mass : 3700 kg kg g g O2.05. Completion of pre-flight portion of fuel log Given:maximum allowable take-off mass 64 400 kg maximum landing mass 56 200 kg maximum zero fuel mass 53 000 kg dry operating mass 35 500 kg estimated load 14 500 kg estimated trip fuel 4 900kg minimum take-off fuel 7 400 kg Find the maximum allowable take-off fuel: kg 0 kg 0 kg

33.01.03. Flight monitoring and in-flight replanning

33.01.03.01. In-flight fuel computations

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or taxi is
e same,
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- a 12 minutes.
- **b** 63 minutes.
- c 44 minutes.

d 4 minutes.

72	For a planned flight the calculated fuel is as follows: Flight time: 3h06min The
id 2536	reserve fuel, at any time, should not be less than 30% of the remaining trip fuel.
I	Taxi fuel: 8 kg Block fuel: 118 kg How much fuel should remain after 2 hours flight
	time?

- a 27 kg trip fuel and 12 kg reserve fuel.
- **b** 39 kg trip fuel and 12 kg reserve fuel.

c 30 kg trip fuel and 9 kg reserve fuel.

d 39 kg trip fuel and no reserve fuel.

- **73** For a planned flight the calculated fuel is as follows: Flight time: 2h42min The reserve fuel, at any time, should not be less than 30% of the remaining trip fuel. Taxi fuel: 9 kg Block fuel: 136 kg How much fuel should remain after 2 hours flight time?
- a 33 kg trip fuel and no reserve fuel.
- **b** 33 kg trip fuel and 10 kg reserve fuel.
- c 23 kg trip fuel and 10 kg reserve fuel.
- d 25 kg trip fuel and 8 kg reserve fuel.

33.01.03.02. Calculation of actual consumption rate

- 74 A VFR flight planned for a Piper Seneca III. At a navigational checkpoint the
- id 2042 remaining usable fuel in tanks is 60 US gallons. The alternate fuel is 12 US gallons. According to the flight plan the remaining flight time is 1h35min. Calculate the highest rate of consumption possible for the rest of the trip.
- a 37.9 US gallons/hour
- b 33.0 US gallons/hour
- c 30.3 US gallons/hour

d 21.3 US gallons/hour

- 75 A multi engine piston aeroplane is on an IFR flight. The fuel plan gives a trip fuel of
 ²⁰⁶⁴ 65 US gallons. The alternate fuel, final reserve included, is 17 US gallons.
 Contingency fuel is 5% of the trip fuel. The usable fuel at departure is 93 US
 gallons. At a certain moment the fuel consumed according to the fuel gauges is 40
 US gallons and the distance flown is half of the total distance.
- a At the destination there will still be 30 US gallons in the tanks
- **b** The remaining fuel is not sufficient to reach the destination with reserves intact
- c At departure the reserve fuel was 28 US gallons
- d At destination the required reserves remain intact.

76 The fuel burn of an aircraft turbine engine is 220 l/h with a fuel density of 0,80. If the density is 0,75, the fuel burn will be:

- a 235 l/h
- **b** 206 l/h
- **c** 220 l/h
- **d** 176 l/h

33.01.03.04. In-flight replanning in case of problems

77 Minimum planned take-off fuel is 160 kg (30% total reserve fuel is included). Assume the groundspeed on this trip is constant. When the aeroplane has done half the distance the remaining fuel is 70 kg. Is diversion to a nearby alternate necessary?

- **a** Diversion to a nearby alternate is not necessary, because it is allowed to calculate without reserve fuel.
- **b** Diversion to a nearby alternate is not necessary, because the reserve fuel has not been used completely.
- c Diversion to a nearby alternate is necessary, because the remaining fuel is not sufficient.
- **d** Diversion to a nearby alternate is necessary, unless the captain decides to continue on his own responsability.
- 78 After flying for 16 min at 100 kt TAS with a 20 kt tail wind component, you have to
- id 5542 return to the airfield of departure. You will arrive after:
- **a** 20 min
- b 24 min
- c 10 min 40 sec
- **d** 16 min

33.01.04. Radio communication and navigation aids

33.01.04.01. Communication frequencies

79 During a flight at night a position has to be reported to ATC. The aeroplane is at a distance of 750 NM from the groundstation and at flight level 350. The frequency to be used is:

- a 5649 kHz.
- **b** 11336 kHz.
- c 17286 kHz.
- d 123.9 MHz.

80 Give the name and frequency of the Flight Information Service for an aeroplane in position (47°59'N, 010°14'E).

a MÜNCHEN INFORMATION 120.65 MHz

b MÜNCHEN INFORMATION 126.95 MHz

c FRANKFURT INFORMATION 128.95 MHz

d MEMMINGEN INFORMATION 122.1 MHz

- 81 Give the frequency of STUTTGART ATIS.
- **id** 4398
- a 135.775 MHz
- b 126.125 MHz
- c 112.250 MHZ
- d 126.125 kHz

82 Give the frequency of ZÜRICH VOLMET.

- **id** 4570
- a 128.525 MHz
- **b** 127.20 kHz
- c 127.20 MHz
- **d** 118.10 MHz

33.01.04.02. Radio navigation and approach aids

83 Refer to the appropriate chart in the Student Pilot Route Manual: Which navigation aid is located in position 48°55'N, 009°20'E ?

- a VOR
- **b** NDB
- c TACAN
- d VOR/DME

84	Refer to the appropriate chart in the Student Pilot Route Manual: Which navigation
id 4572	aid is located in position 48°23'N, 008°39'E?

- a VOR
- **b** NDB
- c VOR/DME
- d VORTAC

85 Refer to the appropriate chart in the Student Pilot Route Manual: Which navigation aid is located in position 48°30'N, 007°34'E?

- a VOR/DME
- **b** NDB
- $\boldsymbol{c} \ \text{VOR}$
- d TACAN

33.02. ICAO ATC FLIGHT PLAN

33.02.01. Types of flight plan

33.02.01.01. ICAO flight plan

	86	A repetitive flight plan (RPL) is filed for a scheduled flight: Paris-Orly to Angouleme,	
id	68	Paris Orly as alternate. Following heavy snow falls, Angouleme airport will be	
		a re-routing of that flight to Limoges.	
а	The p	ilot-in-command must advise ATC of his intention to divert to Limoges at least 15 minutes	
	before	e the planned time of arrival.	
b	The a least l	irline's "Operations " Department has to tansmit a change in the RPL at the ATC office, at nalf an hour before the planned time of departure.	
С	It is no (scheo	ot possible to plan another destination and the flight has to be simply cancelled that day duled flight and not chartered).	
d	The F	RPL must be cancelled for that day and a specific flight plan has to be filed.	
	87	Which of the following statements regarding filing a flight plan is correct?	
id	957		
а	A flyir	ng college can file repetitive flight plan for VFR flights.	
b	Any fl	ight plan should be filed at least 10 minutes before departure.	
С	A fligh	nt plan should be filed when a national FIR boundary will be crossed.	
d	In cas	se of flow control the flight plan should be filed at least three hours in advance of the	
	time of departure.		
id	88 1399	A "current flight plan" is a :	
id a	88 ¹³⁹⁹ flight p ATC.	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and	
^{id} a b	88 ¹³⁹⁹ flight p ATC. filed fl	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan.	
^{id} a b c	88 1399 flight r ATC. filed fl flight r	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure.	
^{id} a b c d	88 1399 flight p ATC. filed fl flight p	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included.	
id a b c d	88 1399 flight p ATC. filed fl flight p filed 1 89	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating	
id a b c d id	88 1399 flight p ATC. filed fl flight p filed fl 89 5650	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is :	
id a b c d id	88 1399 flight r ATC. filed fl flight r filed fl 69 5650 mediu	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is : um plus "M+"	
id a b c d id a b	88 1399 flight p ATC. filed fl flight p filed fl 89 5650 mediu heavy	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is : um plus "M+" //medium "H/M"	
id a b c d id a b c	88 1399 flight p ATC. filed fl flight p filed fl 69 5650 mediu heavy mediu	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is : um plus "M+" //medium "H/M" um "M"	
id a b c d id a b c d	88 1399 flight p ATC. filed fl flight p filed fl filed fl 89 5650 mediu heavy mediu heavy	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is : um plus "M+" //medium "H/M" um "M" / "H"	
id a b c d id a b c d	88 1399 flight p ATC. filed fl flight p filed fl flight p 5650 mediu heavy mediu heavy 90	A "current flight plan" is a : blan in the course of which radio communication should be practised between aeroplane and ight plan. blan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is : um plus "M+" //medium "H/M" um "M" / "H" For the purposes of Item 9 (Wake turbulence category) of the ATS flight plan, an	
id a b c d id a b c d id	88 1399 flight p ATC. filed fl flight p filed fl flight p filed fl 89 5650 mediu heavy mediu heavy 90 5651	A "current flight plan" is a : plan in the course of which radio communication should be practised between aeroplane and ight plan. plan with the correct time of departure. flight plan with amendments and clearance included. An aircraft has a maximum certificated take-off mass of 137000 kg but is operating at take-off mass 135000 kg. In Item 9 of the ATS flight plan its wake turbulence category is : um plus "M+" //medium "H/M" um "M" / "H" For the purposes of Item 9 (Wake turbulence category) of the ATS flight plan, an aircraft with a maximum certificated take-off mass of 62000 kg is :	

- b heavy "H"
- c light "L"
- d medium "M"

	91	When completing an ATS flight plan, an elapsed time (Item 16) of 1 hour 55
id	5654	minutes should be entered as :
а	0115	
b	1H55	
С	115M	
d	0155	
id	92 5655	When completing an ATS flight plan for a European destination, clock times are to
а	Centra	al European Time
b	Local	mean time
с	local s	standard time
d	UTC	
	93	In the ATS flight plan, for a non-scheduled flight which of the following letters
id	5656	schould be entered in Item 8 (Type of Flight) :
а	Ν	
b	N/S	
С	G	
d	Х	
id	94 5658	In the ATS flight plan item 15, it is necessary to enter any point at which a change of cruising speed takes place. For this purpose a "change of speed" is defined as :
	20 km	per hour or 0.1 Mach or more
u h	10 %	TAS or 0.05 Mach or more T_{A}
u c	5% T	AS or 0.01 Mach or more
d	20 kn	ots or 0.05 Mach or more
	95	In the ATS flight plan item 15, when entering a route for which standard departure
id	5659	(SID) and standard arrival (STAR) procedures exist :
а	both	should be entered in the ATS plan where appropriate
b	SIDs	should be entered but not STARs
С	STAR	S should be entered but not SIDs
d	neithe	r SID nor STAR should be entered
id	96 5662	When completing an ATS flight plan for a flight commencing under IFR but possibly changing to VFR, the letters entered in Item 8 (FLIGHT RULES) would be :
а	Х	
b	N/S	
с	G	
d	Y	
	97	In the ATS flight plan Item 19, if the number of passengers to be carried is not
id	5663	known when the plan is ready for filing :
а	the pla	an should be filed with the relevant box blank
b	"TBN	" (to be notified) may be entered in the relevant box
С	an est	timate may be entered but that number may not subsequently be exceeded
d	the pla	an may not be filed until the information is available

9 id 5	8 5664	In an ATS flight plan Item 15, in order to define a position as a bearing and distance from a VOR, the group of figures should consist of :
a ∖	VOR	dent, magnetic bearing and distance in kilometres
b∖	VOR	dent, true bearing and distance in kilometres
c١	VOR	ident, magnetic bearing and distance in nautical miles
d f	ⁱ ull na	me of VOR, true bearing and distance in kilometres
9	9	An aircraft plans to depart London at 1000 UTC and arrive at Munich (EDDM) at
id 5	5665	1215 UTC. In the ATS flight plan Item 16 (destination/EET) should be entered with :
аE	EDDN	1 1215
ЬE	EDDN	1 1415
сE	EDDN	l 0215
d E	EDDN	1 2H15
10	00	In an ATS flight plan Item 15 (route), in terms of latitude and longitude, a significant
id 5	5666	point at 41°35' north 4°15' east should be entered as :
a١	N0413	35E0415
b 4	41°35	N 04° 15'E
C 4	4135N	
d	N4135	5 E00415
10 id 5	D1 5667	In an ATS flight plan, Item 15 (route), a cruising pressure altitude of 32000 feet would be entered as :
аF	FL320	
bF	F320	
cS	S3200	
d 3	32000	
10)2	When an ATS flight plan is submitted for a flight outside designated ATS routes,
id 5	8000	points included in Item 15 (route) should not normally be at intervals of more than :
a 2	20 mi	nutes flying time or 150 km
Di	30 mi	nutes flying time or 370 km
C 1	15 mi	nutes light time or 500 km
u 10		In the ATS flight plan Item 15, a cruicing around of 470 knote will be entered as i
id 5	J3 5669	In the ATS hight plan item 15, a cruising speed of 470 knots will be entered as .
a١	V470	
b∤	KN47()
c (0470k	
d I	N0470	
10)4	In the ATS flight plan Item 13, in a flight plan submitted before departure, the
id 5	5670	departure time entered is the :
аe	estim	ated off-block time

- **b** estimated time over the first point en route
- c estimated take-off time
- d allocated slot time

1	05	In the ATS flight plan Item 15 (Cruising speed), when not expressed as a Mach
id	5671	number, cruising speed is expressed as :
а	IAS	
b	TAS	
С	CAS	
d	Grour	ldspeed
1 id	06	For a repetitive flight plan (RPL) to be used, flights must take place on a regular basis on at least :
а	20.00	rasions
h	10 00	rasions
ĉ	30.00	rasions
d d	50 00	casions
	0000	In the ATC flight plan Item 10 (equipment), the latter to indicate the corrigge of a
id	5673	serviceable transponder - mode A (4 digits-4096 codes) and mode C, is :
а	В	
b	С	
С	А	
d	Р	
3	3.02	.02. Completing the flight plan
33	3.02.	02.01. Information for flight plan obtained from
1	80	An aeroplane is flying from an airport to another. In cruise, the calibrated airspeed
id	67	is 150kt, true airspeed 180 kt, average groundspeed 210 kt, the speed box on the filed flight plan shall be filled as follows:
а	K0150)
b	K0210)
с	K0180)
d	N0180	
1	09	On a flight plan you are required to indicate in the box marked "speed" the planned
id	554	speed for the first part of the cruise or for the entire cruise. This speed is:
а	The e	stimated ground speed
b	The e	quivalent airspeed
С	The ir	ndicated airspeed
d	The t	rue airspeed
1	10	The navigation plan reads: Trip fuel: 100 kg Flight time: 1h35min Taxi fuel: 3 kg
id	2534	Block fuel: 181 kg The endurance on the ICAO flight plan should read:
а	1h 35	min
b	2h 49	min

- **c** 2h 04min
- **d** 2h 52min

1 id	11 2535	The navigation plan reads: Trip fuel: 136 kg Flight time: 2h45min Calculated reserve fuel: 30% of trip fuel Fuel in tank is minimum (no extra fuel on board) Taxi
	I	fuel: 3 kg The endurance on the ICAO flight plan should read:
а	2h49n	nin
b	2h45n	nin
С	3h34r	nin
d	3h38n	nin
1	12	If your destination airport has no ICAO indicator, in the appropriate box of your flight
Ia	5514	plan, you write:
a	////	
b		
C		
a 	10	
T id	13 5525	The cruising speed to write in the appropriate box of a flight plan is:
а	ground	d speed
b	indica	ted air speed
с	true a	ir speed
d	calibra	ated air speed
1	14	In the appropriate box of a flight plan, for endurance, one must indicate the time
Ia	5526	corresponding to:
a L	the to	otal usable fuel on board
D	the re	quired fuel for the flight plue the alternate and 45 minutes
ט ה	the tet	quired fuel for the hight plus the alternate and 45 minutes
u 		The maximum permissible take off mass of an aircraft for the L wake turbulance
id	1 5 5529	category on a flight plan is:
а	7 000	kg
b	2 700	kg
С	5 700	kg
d	10 00) kg
_1	16	In the appropriate box of a flight plan form, concerning equipment, the letter to be
id	5530	used to indicate that the aircraft is equipped with a mode A 4096 codes transponder with altitude reporting capability is :
а	s	
b	P	
С	С	
d	А	
1	17	When a pilot fills in a flight plan, he must indicate the wake turbulence category.
id	5543	This category is a function of which mass?
а	actual	take-off mass
b	estima	ated take-off mass
С	maxin	num certified landing mass

d maximum certified take-off mass

118 In the appropriate box of a flight plan form, corresponding to the estimated time of

- id 5545 departure, the time indicated is that at which the aircraft intends to :
- a start-up
- b take-off
- c go off blocks
- d pass the departure beacon

33.02.03. Filing the flight plan

33.02.03.01. Procedures for filing

119 You have a flight plan IFR from Amsterdam to London. In the flight plan it is noted that you will deviate from the ATS route passing the FIR boundary Amsterdam/London. The airway clearance reads: Cleared to London via flight planned route. Which of the following statements is correct?

a The route according to the flight plan is accepted.

- **b** The filed deviation is not accepted.
- c You will get a separate clearance for the deviation.
- **d** It is not allowed to file such a flight plan.

120 How many hours in advance of departure time should a flight plan be filed in the case of flights into areas subject to air traffic flow management (ATFM)?

- a 1:00 hour.
- b 3:00 hours.
- c 0:30 hours.
- **d** 0:10 hours.

121 For a flight plan filed before the flight, the indicated time of departure is: id 5524

a the estimated off-block time

- **b** the time at which the flight plan is filed.
- c the time of take-off.
- d the time overhead the first reporting point after take-off.

122 It is possible, in flight, to: 1 - file an IFR flight plan 2 - modify an active IFR or VFR flight plan 3 - cancel an active VFR flight plan 4 - close an active VFR flight plan Which of the following combinations contains all of the correct statements?

- **a** 2 3 4
- **b** 1 2 4
- **c** 1 2 3
- d 1 2 3 4

- a 1755 UTC
- **b** 1725 UTC
- c 1745 UTC
- d 1715 UTC

¹²³ The planned departure time from the parking area is 1815 UTC The estimated takeid 5544 off time is 1825 UTC The IFR flight plan must be filed with ATC at the latest at:

33.02.03.03. Requirements of the State

- **124** From the options given below select those flights which require flight plan
- id 69 notification: I Any Public Transport flight. 2 Any IFR flight 3 Any flight which is to be carried out in regions which are designated to ease the provision of the Alerting Service or the operations of Search and Rescue. 4 - Any cross-border flights 5 -Any flight which involves overflying
- **a** 1+5
- b 2+4
- **c** 1+2+3
- **d** 3+4+5

33.02.04. Closing the flight plan

33.02.04.01. Responsibilities and procedures

- 125 ^{id} ⁶⁶ If a pilot lands at an aerodrome other than the destination aerodrome specified in ^{id} ⁶⁶ the flight plan, he must ensure that the ATS unit at the destination aerodrome is informed within a certain number of minutes of his planned ETA at destination. This number of minutes is:
- **a** 10
- **b** 15
- c 30
- **d** 45

33.02.05. Adherence to flight plan

33.02.05.01. Tolerances allowed by the State

126 When an ATS flight plan has been submitted for a controlled flight, the flight plan

id 5660 should be amended or cancelled in the event of the off-block time being delayed by :

- a 90 minutes or more
- b 45 minutes or more
- c 60 minutes or more
- d 30 minutes or more

33.02.05.02. In-flight amendment of flight plan

127 An aeroplane is on an IFR flight. The flight is to be changed from IFR to VFR. Is it possible?

a Yes, the pilot in command must inform ATC using the phrase "cancelling my IFR flight".

- **b** No, you have to remain IFR in accordance to the filed flight plan.
- **c** No, only ATC can order you to do this.
- ${\bf d}\,$ Yes, but only with permission from ATC.
- **128** During an IFR flight TAS and time appear to deviate from the data in the flight plan. ¹³⁹⁷ The minimum deviations, that should be reported to ATC in order to conform to PANS-RAC, are:
- **a** TAS 5 kt and time 5 minutes.
- **b** TAS 3% and time 3 minutes.
- c TAS 5% and time 3 minutes.
- d TAS 10 kt and time 2 minutes.

33.03. PRACTICAL FLIGHT PLANNING

33.03.01. Chart preparation

33.03.01.01. Plot tracks and measure directions

- 129 Flying from SAULGAU airport (48°02'N, 009°31'E) to ALTENSTADT airport
- id 4395 (47°50'N, 010°53'E). Find magnetic course and the distance.
- a Magnetic course 102°, distance 82 NM
- **b** Magnetic course 282°, distance 56 NM

c Magnetic course 102°, distance 56 NM

- d Magnetic course 078°, distance 82 NM
- **130** Flying from ERBACH airport (48°21'N, 009°55'E) to POLTRINGEN airport (48°33'N, 008°57'E). Find magnetic course and the distance.
- a Magnetic course 108°, distance 60 NM
- b Magnetic course 252°, distance 41 NM
- c Magnetic course 287°, distance 41 NM
- d Magnetic course 287°, distance 60 NM
- **131** Flying from Position SIGMARINGEN (48°05'N, 009°13'E) to BIBERACH airport (48°07'N, 009°46'E). Find magnetic course and the distance.
- a Magnetic course 086°, distance 32 NM
- b Magnetic course 093°, distance 41 NM
- c Magnetic course 267°, distance 22 NM
- d Magnetic course 086°, distance 22 NM
- **132** On airway PTS P from Vigra (62°334N 006°02'E), the initial great circle grid course is :
- **a** 347
- **b** 353
- c 344
- **d** 350
- **133** On a direct great circle course from Shannon (52°43' N 008°53'W) to Gander id 5726 (48°54'N054°32'W), the (a) average true course, and (b) distance, are :
- a a) 244° (b) 1520 NM
- **b** a) 281° (b) 2730 NM

c (a) 262° (b) 1720 NM

- **d** a) 281° (b) 1877 NM
- **134** The initial great circle true course from Keflavik (64°00'N 022°36' W) to Vigra id 5727 (62°33'N 006°02'E) measures 084°. On a polar enroute chart where the grid is aligned with the 000° meridian the initial grid course will be :
- **a** 096°
- **b** 080°
- c 106°
- **d** 066°

135The initial great circle course from position A (80°00'N 170°00'E) to position Bid57285728(75°00'N 011°E) is 177° (G). The final grid course at position B will be :

- **a** 172° (G)
- **b** 194° (G)
- c 177° (G)
- **d** 353° (G)

33.03.02. Navigation plans

33.03.02.01. Completing the navigation plan using:

- **136** A descent is planned from 7500 ft MSL so as to arrive at 1000 ft MSL 6 NM from a VORTAC. With a GS of 156 kts and a rate of descent of 800 ft/min. The distance from the VORTAC when descent is started is :
- a 15,0 NM
- b 27,1 NM
- **c** 11,7 NM
- **d** 30,2 NM

137 A sector distance is 450 NM long. The TAS is 460 kt. The wind component is 50 kt tailwind. What is the still air distance?

- a 414 Nautical Air Miles (NAM)
- **b** 499 Nautical Air Miles (NAM)
- c 406 Nautical Air Miles (NAM)
- d 511 Nautical Air Miles (NAM)
- **138** The still air distance in the climb is 189 Nautical Air Miles (NAM) and time 30
- id 2770 minutes. What ground distance would be covered in a 30 kt head wind?
- a 188 NM
- **b** 203 NM
- c 174 NM
- **d** 193 NM

33.03.03. Simple fuel plans

33.03.03.01. Preparation of fuel logs

- (For this question use Flight Planning Manual MEP 1 Figure 3.2) A flight is to be made in a multi engine piston aeroplane (MEP1). The cruising level will be 11000ft. The outside air temperature at FL is -15 ° C. The usable fuel is 123 US gallons. The power is set to economic cruise. Find the range in NM with 45 min reserve fuel at 45 % power.
- a 752 NM
- **b** 852 NM
- c 610 NM
- **d** 602 NM

140 (For this Question use Flight Planning & Monitoring MEP1 Fig. 3.5) Given: FL 75 ^{id} ²⁵⁶⁴ Lean mixture Economy Powersetting Find: Endurance in hours with no reserve

- **a** 06:12
- b 05:01
- **c** 06:06
- **d** 05:11

- 141 (For this Question use Flight Planning & Monitoring SEP1) A flight has to be made
 id 4574 with the single engine sample aeroplane. For the fuel calculation allow 10 lbs fuel
 for start up and taxi, 3 minutes and 1 gallon of additional fuel to allow for the climb,
 10 minutes and no fuel correction for the descent. Planned flight time (overhead to overhead) is 03 hours and 12 minutes. Res
- a 283 lbs
- **b** 268 lbs
- **c** 252 lbs
- **d** 215 lbs
- (For this Question use Flight Planning & Monitoring SEP1) A flight has to be made with the single engine sample aeroplane. For the fuel calculation allow 10 lbs fuel for start up and taxi, 3 minutes and 1 gallon of additional fuel to allow for the climb, 10 minutes and no fuel correction for the descent. Planned flight time (overhead to overhead) is 02 hours and 37 minutes. Res
- a 250 lbs
- **b** 208 lbs
- **c** 270 lbs
- d 265 lbs

33.04. IFR (AIRWAYS) FLIGHT PLANNING

33.04.01. Meteorological considerations

Which describes the worst hazard, if any, that could be associated with the type of 143 id 5210 | feature at 37.7°N 015°E ? a Reduced visibility b Severe attenuation in the HF R/T band c Engine flame out and windscreen damage d There is no hazard The surface weather system over England (53°N 002°W) is 144 **id** 5211 a a warm front moving southeast **b** a depression moving north c an occluded front moving east d a cold front moving east 145 In the vicinity of PARIS (49°N 003°E) the tropopause is at about **id** 5212 a FL350 **b** FL340 **c** FL400 **d** FL380 146 Which describes the maximum intensity of icing, if any, at FL110 in the vicinity of id 5213 CASABLANCA (33°N 008°W) ? a Light **b** Moderate c Severe d Nil Which best describes the significant cloud, if any, forecast for the area southwest 147 id 5214 of BODO (67°N 014°E) a Nil b 5 to 7 oktas AC, base FL100, tops FL180 c 3 to 7 oktas AC, base below FL100, tops FL180 d 5 to 7 oktas AC, base below FL100, tops FL180 Which best describes be maximum intensity of icing, if any, at FL150 in the vicinity 148 id 5215 (west) of BUCHAREST (45°N 026°E) ?

a Nil

- **b** Light
- c Severe
- d Moderate

1 id	49 5216	Which best describes the maximum intensity of CAT, if any, forecast for FL330 over BENGHAZI (32°N 020°E) ?
а	Nil	
b	Light	
с	Mode	rate
d	Severe	9
1	50	The maximum wind velocity (°/kt) shown in the vicinity of MUNICH (48°N 012°E) is :
id	5217	
а	260/13	30
b	080/13	30
С	260/16	60
d	290/23	30
1	51	The wind velocity over GERMANY is
id	5218	
а	130 kt	at FL320 maximum velocity not shown on chart
b	130 kt	at FL340 maximum velocity not shown on chart
С	a max	imum of 230 kt at FL 320
d	a max	kimum of 130 kt at FL320
1	52	The wind direction and velocity (°/kt) at 50°N 040°E is:
Id	5219 170/26	-
d 6	250/25	
u o	200/20	
с А	200/20	
	500/50	
1 id	53	The wind direction and velocity (°/kt) at 60°N 015°E is
a	200/20	
b	020/20	
c	210/25	5
d	030/25	5
1	54	What is the mean temperature deviation (°C) from the ISA over $50^{\circ}N$ $010^{\circ}W$ 2
id	54 5222	
а	+/-0	
b	+15	
С	+9	
d	-5	
1	55	The wind direction and velocity (°/kt) at 40°N 040°E is
id	5223	
а	280/20)
b	330/60)
С	150/20)
d	330/20)

1 id	56 5228	A METAR reads : SA1430 35002KY 7000 SKC 21/03 QI024 = Which of the following information is contained in this METAR ?
а	temp	erature/dewpoint
b	runwa	y in use
с	day/m	onth
d	period	of validity
1	57	What mean temperature (°C) is likely on a course of 360° (T) from 40°N to 50°N at
id	5229	040°E ?
а	-19	
b	-23	
С	-21	
d	-56,5	
1 id	58 5230	Which of the following flight levels, if any, is forecast to be clear of significant cloud, icing, turbulence, and CAT along the marked route from SHANNON (53°N 10°W) to BERLIN (53°N 13°E) ?
а	FL250	
b	FL 210)
С	FL290	
d	None	
1 id	59	The W/V (°/kt) at 50°N015°W is:
a	320/40	
b	140/40	
c	320/25	5
d	140/25	5
1	60	What mean temperature (°C) is likely on a true course of 270° from 025° E to
id	5687	010°E at 45°N ?
а	-5	
b	-25	
С	-15	
d	-30	
1	61	The W/V (°/kt) at 40°N 020°W is
id o	5688	
a h	331/10	
C C	135/4	
d	350/60	
1	62	What is the temperature deviation (°C) from ISA over 50° N 010°E ?
id	5689	
а	-55	
b	+8	
С	+2	

1	63	The W/V (°/kt) at 60° N015° W is
ld Q	5690 250/4	
d h	250/10	
D C	270/10	
с А	000/10	
	030/10	
1 id	64 5691	The approximate mean wind component (kt) along true course 180° from 50°N to 40°N at 005° W is
а	headv	vind 20 kt
b	tail wi	nd 25 kt
С	tail wi	nd 15 kt
d	head	wind 15 kt
1 id	65 5693	Which best describes the maximum intensity of icing, if any, at FL160 in the vicinity of BUDAPEST?
а	mode	rate
b	severe	
С	light	
d	nil	
1 id	66 5694	Which describes the intensity of turbulence, if any, at FL 150 in the vicinity of ALGIERS?
а	mode	rate
b	mode	erate or severe
b c	mode light	erate or severe
b c d	mode light nil	erate or severe
b c d 1 id	mode light nil 67	The surface system west of PARIS is a
b c d 1 id a	mode light nil 67 5695 cold f	The surface system west of PARIS is a
b c d 1 id a b	mode light nil 67 5695 cold f warm	The surface system west of PARIS is a front moving southeast front moving north
b c d 1 id a b c	mode light nil 67 5695 cold f warm statior	The surface system west of PARIS is a front moving southeast front moving north hary occluded front
b c d 1 id a b c d	mode light nil 67 5695 cold f warm station cold fr	The surface system west of PARIS is a front moving southeast front moving north hary occluded front ront moving west
b c d 1 id a b c d 1	mode light nil 67 5695 cold f warm statior cold fr	The surface system west of PARIS is a iront moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL
b c d 1 id b c d 1 id	mode light nil 67 5695 cold f warm statior cold fr 68 5696	The surface system west of PARIS is a front moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL
b c d 1 id a b c d 1 id a	mode light nil 67 5695 cold f station cold fr 68 5696 300	The surface system west of PARIS is a front moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL
b c d 1 id a b c d 1 id a b	mode light nil 67 5695 cold f warm statior cold fr 5696 300 270	The surface system west of PARIS is a front moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL
b d 1 id a b c d 1 id a b c c	mode light nil 5695 cold f warm statior cold fr 5696 300 270 250	The surface system west of PARIS is a iront moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL
b d l id a b c d l id a b c d l id a b c d l id a b c d	mode light nil 67 5695 cold f warm statior cold fr 68 5696 300 270 250 290	The surface system west of PARIS is a front moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL
b d 1 id a b c d 1 id a b c d 1 id a b c d	mode light nil 67 5695 cold f warm statior cold fr 68 5696 300 270 250 290	The surface system west of PARIS is a Tont moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL Which best describes the significant cloud forecast over TOULOUSE (44°N001°E)
b d 1 id b c d 1 id a b c d 1 id a b c d 1 id a b c d 1 id a b c d 1 id a b c d	mode light nil 5695 cold f warm statior cold fr 68 5696 300 270 250 290	The surface system west of PARIS is a ront moving southeast front moving north hary occluded front ront moving west In the vicinity of GLASGOW the tropopause is at about FL Which best describes the significant cloud forecast over TOULOUSE (44°N001°E) ?
b c d 1 id a	mode light nil 67 5695 cold f warm statior cold fr 68 5696 300 270 250 290 69 5697 broke	The surface system west of PARIS is a Toot moving southeast front moving north hary occluded front cont moving west In the vicinity of GLASGOW the tropopause is at about FL Which best describes the significant cloud forecast over TOULOUSE (44°N001°E) ? m/overcast layer, base below FL100 tops FL180, moderate icing
b c d <mark>1</mark> id a b c d 1 id a b	mode light nil 5695 cold f warm statior cold fr 68 5696 300 270 250 290 69 5697 broke	The surface system west of PARIS is a ront moving southeast front moving north hary occluded front ont moving west In the vicinity of GLASGOW the tropopause is at about FL Which best describes the significant cloud forecast over TOULOUSE (44°N001°E) ? Which best describes the significant cloud forecast over TOULOUSE (44°N001°E) ?
b c d <mark> 1</mark> id a b c d <mark> 1</mark> id a b c d <mark> 1</mark> id a b c	mode light nil 67 5695 cold f warm statior cold fr 68 5696 300 270 250 290 69 5697 broke well so	The surface system west of PARIS is a Tont moving southeast front moving north hary occluded front ont moving west In the vicinity of GLASGOW the tropopause is at about FL Which best describes the significant cloud forecast over TOULOUSE (44°N001°E) ? Which best describes the significant cloud forecast over TOULOUSE (44°N001°E) ? Provercast layer, base below FL100 tops FL180, moderate icing eparated CB base FL100 tops to FL 180 hovercast layer, base below FL100 tops FL180, severe icing

170 Which describes the maximum intensity of turbulence, if any, forecast for FL260 over TOULOUSE (44°N001°E) ?

- a moderate
- **b** severe
- c light
- **d** nil

171 Over LONDON (51°N000°E/W), the lowest FL listed which is unaffected by CAT is: id 5700

- **a** 360
- **b** 390
- **c** 250
- d 220

33.04.02. Selection of routes

33.04.02.01. Preferred airways routings

172 (For this question use Route Manual chart E(HI)4&5) An aeroplane has to fly from

- id 2054 Salzburg (48°00.2'N 012°53.6'E) to Klagenfurt (46°37.5'N 014°33.8'E). Which statement is correct ?
- a The airway UB5 is closed for southbound traffic above FL 200.
- **b** The airway UB5 cannot be used, there is one way traffic to the north.

c The airway UB5 can be used for flights to/from Klagenfurt and Salzburg.

d The airway UB5 is closed in this direction except during the weekends.

173Of the following, the preferred airways routing from FRANKFURT FFM 114.2id5049(50°03' N008°38'E) to KOKSY (51°06'N 002°39'E) above FL245, on a Wednesdayis :

a UR10 NTM UB6 BUB ATS

- b UG1
- c UB69 DINKI UB6 BUB ATS
- d UG108 SPI UG1

174 Of the following, the preferred airways routing from MARTIGUES MTG 117.3 id 5050 (43°23'N 005°05'E) to ST PREX SPR 113.9 (46°28'N 006°27'E) above FL245 is :

- **a** UB28
- b UB284 VILAR UB28
- c UB282 DGN UB46
- d UA6 LSA UG52

175 Of the following, the preferred airways routing from AMBOISE AMB 113.7 (47°26'N id 5051 001°04'E) to AGEN AGN (43°53°'N 000°52'E) above FL200 is:

- a UH40 FOUCO UH20 PERIC UA34
- **b** UB19 POI UB195
- c UA34
- d UB19 CGC UA25

- **176** Of the following, the preferred airways routing from CLACTON CLN 114.55
- id 5052 (51°51'N 001°09'E) to DINARD DIN 114.3 (48°35'N 002°05'W) above FL245 is:

a UB29 LAM UR1 ORTAC UR14

- b UR12 MID UA47 DPE UA475 SOKMU UH111
- c UR12 MID UR8 SAM UB11 BARLU UW115
- d UB29 LAM UR1 MID UA34 LILAN UR9

33.04.02.02. Extraction of tracks and distances

177 (For this question use Route Manual chart E(HI)4) An aeroplane has to fly from ¹d ²⁰⁵² Abbeville (50°08.1'N 001°51.3'E) to Biggin (51°19.8'N 000°00.2'E). What is the distance of this leg ?

- **a** 62NM
- **b** 38 NM
- **c** 64NM
- d 100 NM
- 178(For this question use Route Manual chart E(HI)4) An aeroplane has to fly fromid2053Abbeville (50°08.1'N 001°51.3'E) to Biggin (51°19.8'N 00°00.2'E). At Biggin you canfind : 141°. This is :
- **a** The average true course of the great circle from Biggin to Abbeville.
- ${\bf b}\,$ The magnetic course to fly inbound to Biggin.

c The magnetic great circle course from Biggin to Abbeville.

- ${\bf d}\,$ The radial, referenced to true north, of Biggin to fly inbound.
- **179** (For this question use Route Manual chart E(HI)4) Planning a IFR flight from Paris ¹²³²⁹ Charles de Gaulle (N49 00.9 E002 36.9) to London Heathrow (N51 29.2 W000 27.9). Find the average true course from Paris to London.
- **a** 330°.
- **b** 142°.
- c 322°.
- **d** 343°.
- (For this question use Route Manual chart E(HI)4) Planning a IFR flight from Paris
 ^{1d} 2330 Charles de Gaulle (N49 00.9 E002 36.9) to London Heathrow (N51 29.2 W000 27.9). Determine the preplanning distance by calculating the direct distance plus 10%. The preplanning distance is:
- **a** 188 NM.
- b 207 NM.
- **c** 308 NM.
- d 218 NM.
- Planning an IFR-flight from Paris to London (Heathrow). Assume: STAR is BIG 2A,
 Variation 5° W, en-route TAS 430 kts, W/V 280/40, descent distance 76NM.
 Determine the magnetic course, ground speed and wind correction angle from ABB 116.6(N50 08.1 E001 51.3) to top of descent.
- a MC 141°, GS 396 kt, WCA -3°
- **b** MC 141°, GS 396 kt, WCA +3°
- c MC 319°, GS 396 kt, WCA -3°
- d MC 321°, GS 396 kt, WCA -3°

1	82	Planning an IFR-flight from Paris to London. Determine the distance of the
id	4562	departure route ABB 8A.

- a 72.5 NM
- b 74.5 NM
- c 56 NM
- **d** 83 NM

183 Planning an IFR-flight from Paris (Charles de Gaulle) RWY 27 to London. Given:
 ^{id} 4563 Distance from PARIS Charles-de-Gaulle to top of climb 50 NM Determine the distance from the top of climb (TOC) to ABB 116.6.

- a 36.5 NM
- **b** 33 NM
- c 24.5 NM
- **d** 31 NM
- 184 Planning an IFR-flight from Paris to London (Heathrow) via initial approach fix (IAF)
 id 4567 Biggin VOR . Given: distance from top of descent (TOD) to Rwy 27R is 76 NM
 Determine the distance from ABB 116.6 to TOD.
- a 49 NM
- **b** 60 NM
- **c** 100 NM
- **d** 55 NM
- 185 Planning an IFR-flight from Paris Charles de Gaulle to London. SID is ABB 8A.
 ^{id} 4569 Assume Variation 3° W, TAS 430kts, W/V 280/40 and distance to top of climb 50NM Determine the magnetic course, ground speed and wind correction angle from top of climb to ABB 116.6.
- a MC 169°, GS 450 kt, WCA +4°
- **b** MC 169°, GS 414 kt, WCA +5°
- c MC 349°, GS 414 kt, WCA +5°
- d MC 349°, GS 414 kt, WCA -5°

186The magnetic course/distance from DINKELSBUHL DKB 117.8 (49°09'N010°14'E)id5053to ERLANGEN ERL 114.9 (49°39'N011°09'E) on airway UR11 is;

- a 050°/47 NM
- **b** 230°/97NM
- c 133°/85 NM
- d 052°/97 NM

187The magnetic course/distance from GROSTENQUIN GTQ 111.25 (49°00'Nid5054006°43'E) to LINNA (49°41'N 006°15'E) on airway R7 is:

- a 157°/58 NM
- b 337°/46 NM
- c 337°/31 NM
- **d** 337°/58 NM

188 The magnetic course/distance from ELBE LBE 115.1 (53°39'N 009°36'E) to ^{id} ⁵⁰⁵⁵ LUNUD (54°50'N 009°19'E) on airway H12 is:

- a 339°/80 NM
- b 352°/96 NM
- c 352°/72 NM
- d 339°/125 NM

l id	89 5056	The initial magnetic course/distance from EELDE EEL 112.4 (53°10'N 006°40'E) to WELGO (54°18'N 007°25'E) on airway A7 is:
а	024°/	73 NM
b	023°/	73 NM
С	024°/	20 NM
d	024°/	47 NM
1	90	The magnetic course/distance from CAMBRAI CMB 112.6 (50°14'N 003°09'E) to
id	5057	TALUN (49°33'N 003°25'E) on airway B3 is:
а	169°/6	58 NM
b	349°/2	26 NM
С	169°/4	12 NM
d	349°/4	12 NM
1 id	91 5058	The magnetic course/distance from WALLASEY WAL 114.1 (53°23N 003°28'W° to LIFFY (53°29'N 005°30'W) on airway B1 is:
а	279°/′	14 NM
b	279°/8	35 NM
С	311°/′	14 NM
d	311%	35 NM
1 id	92 5059	The magnetic course/distance from TRENT TNT 115.7 (53°03'N 001°40'W) to WALLASEY WAL 114.1 (53°23'N 003°08W) on airway VR3 is:
а	297°/7	70 NM
b	117°/5	57 NM
С	297°/5	57 NM
d	117%	74 NINA
1	117 /1	
id	193 5060	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is:
id a	193 5060 007°/6	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM
^{id} a b	193 5060 007°/6 052°/5	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM
^{id} a b c	193 5060 007°/6 052°/6 105°/	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM
id a b c d	93 5060 007°/6 052°/5 105°/7 132°/2	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM
id a b c d 1	193 5060 007°/¢ 052°/ ¢ 105°/ ² 132°/ ² 194	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to
id a b c d 1 id	193 5060 007°/0 052°/2 105°/ ² 132°/2 194 5061	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 60 NM 60 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is:
id b c d 1 id a	193 5060 007°/6 052°/5 105°/7 132°/4 194 5061 061°/2 048°/7	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is: 28 NM
id a b c d 1 id a b c	105°/′ 105°/′ 105°/′ 132°/⁄ 194 5061 061°/⁄ 048°/⁄	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is: 28 NM 16 NM
id a b c d d 1 id a b c ל	105°/ 105°/ 105°/ 132°/ 132°/ 132°/ 132°/ 061°/2 048°/ 041°/2	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is: 28 NM 16 NM
id a b c d 1 id a b c d c d	193 5060 007°/6 052°/ 105°/ 132°/ 132°/ 194 5061 061°/2 048°/4 061°/2 041°/7 041° /7	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 60 NM 60 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is: 28 NM 16 NM 87 NM 78 NM
id a b c d 1 id a b c d 1 id id	107°/0 007°/0 052°/2 105°/2 132°/2 132°/2 194 5061 061°/2 048°/2 048°/2 041°/7 195 5062	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 30 NM 50 NM 50 NM 50 NM 53 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is: 28 NM 66 NM 57 NM 78 NM The magnetic course/distance from SALZBURG SBG 113.8 (48°00'N 012°54'E) to STAUB (48°44'N 012°38'E) on airway UB5 is:
id b c d 1 id b c d 1 id a b c d 1 id a	193 5060 007°% 052°% 105°% 132°% 105 °% 132°% 061°% 048°% 041°% 195 5062 346°%	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 60 NM 60 NM 60 NM 63 NM 63 NM 75 NM 78 NM 78 NM 78 NM 78 NM 78 NM 75 The magnetic course/distance from SALZBURG SBG 113.8 (48°00'N 012°54'E) to STAUB (48°44'N 012°38'E) on airway UB5 is: 64 NM
id a b c d l id a b c d l id a b	107°/0 007°/0 052°/2 105°/2 132°/2 132°/2 194 5061 061°/2 048°/2 041°/7 195 5062 346°/0 346°/0	The magnetic course/distance from TANGO TGO 112.5 (48°37'N 009°16'E) to DINKELSBUHL DKB 117.8 (49°09'N 010°14E) on airway UR11 is: 50 NM 50 NM 105 NM 13 NM The magnetic course/distance from ST PREX SPR 113.9 (46°28'N 006°27'E) to FRIBOURG FRI 115.1 (46°47'N 007°14'E) on airway UG60 is: 28 NM 16 NM 17 NM 78 NM The magnetic course/distance from SALZBURG SBG 113.8 (48°00'N 012°54'E) to STAUB (48°44'N 012°38'E) on airway UB5 is: 34 NM

d 346°/43 NM

- **196** The magnetic course/distance from ELBA ELB 114.7 (42°44'N 010°24'E) to SPEZI id 5063 (43°49'N 009°34'E) on airway UA35 is:
- a 332°/118 NM
- **b** 152°/42 NM
- c 322°/60 NM
- d 332°/76 NM

197 The magnetic course/distance from LIMOGES LMG 114.5 (45°49'N 001°02'E) to

id 5064 CLERMONT FERRAND CMF 117.5 (45°47'N 003°11'E) on airway UG22 is:

- a 067°/ 122 NM
- b 094°/ 90 NM
- **c** 113°/ 142 NM
- **d** 046°/ 70 NM

33.04.02.03. Frequencies and identifiers

198 (For this question use Route Manual chart E(HI)4) An aeroplane has to fly from ^{id} ²⁰⁵⁶ Salzburg (48°00.2'N 012°53.6'E) to Klagenfurt (46°37.5'N 014°33.8'E). At Salzburg there is stated on the chart D 113.8 SBG. That means :

a VOR/DME with identification SBG frequency 113.8 MHz can be used.

- **b** Only the DME with identification SBG can be used, for which frequency 113.8 MHz should be tuned, VOR is not available.
- c VOR/DME SBG will be deleted in the future and cannot be used for navigation.

d ILS/DME 113.8 MHz of Salzburg airport can be used for navigation.

199 The radio navigation aid at TOPCLIFFE (54°12'N 001°22'W) is a:

id 5065

a TACAN only, channel 84, (frequency 113.7 MHz)

- b TACAN, channel 84, and a VOR frequency 113.7 MHz only
- c TACAN, channel 84, and an NDB frequency 92 kHz only
- d VORTAC, frequency 113.7 MHz, and an NDB frequency 92 kHz

200 The radio navigation aid serving STRASBOURG (48°30'N 007°34'E) is a:

- **id** 5066
- a VOR only, frequency 115.6 MHz

b VOR/TACAN, frequency 115.6 MHz

- c DME only, channel 115.6
- d TACAN only, frequency 115.6 MHz

201 The radio navigation aid at ST DIZIER (48°38N 004°53'E) is a:

id 5067

- a TACAN, channel 87, and NDB frequency 114.0 kHz
- b VOR, frequency 114.0 MHz, and TACAN channel 87

c TACAN, channel 114.0

d TACAN, channel 87, frequency 114.0 MHz

202 The radio navigation aid STAD (51°45'N 004°15'E) is:

id 5069

- a a VOR, frequency 386 MHz
- b an NDB, frequency 386 kHz
- c a VOR/DME, on channel 386
- d a TACAN, on channel 386

203 The radio navigation aid at CHIOGGIA (45°04'N 012°17'E) is a:

id 5070

a VOR/DME, frequency 114.1 MHz, and NDB frequency 408 kHz

- **b** VOR, frequency 114.1 MHz, and TACAN channel 408
- c VOR, frequency 114.1 MHz, and TACAN frequency 408 MHz
- d VOR/DME only, frequency 114.1 MHz

204 The radio navigation aid on airway UG4 at LUXEUIL (47°41'N 006°18'E) is a:

id 5071

a VOR only, identifier LUL

- **b** VOR, identifier LUL, frequency paired with TACAN identifier LXI
- c VOR/DME and NDB, identifier LXI
- **d** VOR/DME only, identifier LUL

205 The radio navigation aid at BELFAST CITY (54°37'N 005°53'W) is :

id 5072

a a TACAN, channel 420

b an NDB, frequency 420 kHz, NOT continuous operation

- **c** a fan marker, frequency 420 kHz
- d an NDB, frequency 420 kHz, continuous operation

206 The radio navigation aid at SHANNON (52°43'N 008°53'W) is :

id 5073

 ${\boldsymbol{a}}~$ an NDB, frequency 352 kHz

b a VOR/DME, frequency 113.3 MHz

- c a TACAN, frequency 113.3 kHz
- **d** a VOR only, frequency 113.3 MHz

207 The VOR and TACAN on airway G9 at OSNABRUCK (52°12'N 008°17'E) are: id 5074

a NOT frequency paired, and have different identifiers

- b frequency paired, and have different identifiers
- c NOT frequency paired, and have the same identifier
- d frequency paired, and have the same identifier

208 The NDB at DENKO (52°49'N 015°50'E) can be identified on:

id 5075

- a Channel 440, BFO on
- b Frequency 440 kHz, BFO on
- c Channel 440, BFO off
- d Frequency 440 kHz, BFO off

209 The airway intersection at RONNEBY (56°18'N 015°16'E) is marked by: id 5076

- a an NDB callsign N
- **b** a TACAN callsign RON
- **c** a fan marker callsign LP
- d an NDB callsign LF

33.04.02.04. Minimum en-route altitudes
210 (For this question use Route Manual chart E(HI)4) An aeroplane has to fly from id 2055 Salzburg (48°00.2'N 012°53.6'E) to Klagenfurt (46°37.5'N 014°33.8'E). Which statement is correct ?
a The minimum enroute altitude (MEA) is 13400 ft.
b The minimum grid safe altitude on this route is 13400 ft above MSL.
c The minimum sector altitude (MSA) is 13400 ft.
d The minimum obstacle clearance altitude (MOCA) on this route is 10800 ft above MSL.
211 Aeroplanes intending to use airway UR14 should cross GIBSO intersection (50°45'N 002°30'W) at or above:
a FL160
b FL140
c FL250
d FL200
212 An airway is marked 3500T 2100 a. This indicates that: id 5082
a the minimum enroute altitude (MEA) is 3500 ft
b the minimum obstruction clearance altitude (MOCA) is 3500 ft
c the airway base is 3500 ft MSL
d the airway is a low level link route 2100 ft - 3500 ft MSL
213 The minimum enroute altitude available on airway UR160 from NICE NIZ 112.4 id 5083 (43°46'N 007°15'E) to BASTIA BTA 116.2 (42°32'N 009°29'E) is:
a FL260
b FL200
c FL210
d FL250
214 The minimum enroute altitude that can be maintained continuously on airway UA34 id 5084 from WALLASEY WAL 114.1 (53°23'N 003°08'W) to MIDHURST MID 114.0 (51°03'N 000°37'W) is :
a FL250
b FL245
c FL290
d FL330
215 An airway is marked FL 80 1500 a. This indicates that:
a 1500 ft MSL is the minimum radio reception altitude (MRA).

- **b** the airway base is 1500 ft MSL.
- c the airways extends from 1500 ft MSL to FL 80.
- d the minimum enroute altitude (MEA) is FL 80.

216 The minimum enroute altitude (MEA) that can be maintained continuously on airway G4 from JERSEY JSY 112.2 (49°13'N 002°03'W) to LIZAD (49°35'N 004°20'W) is :

- a 2800 ft MSL
- **b** FL60
- c FL140
- d 1000 ft MSL

217 An airway is marked 5000 2900a. The notation 5000 is the :

- **id** 5088
- a base of the airway (AGL)
- **b** maximum authorised altitude (MAA)
- c minimum holding altitude (MHA)

d minimum enroute altitude (MEA)

218The minimum enroute altitude that can be maintained continuously on airwayid5089B65/H65 from DOXON (55°27'N 018°10'E) to RONNE ROE 112.0 (55°04'N
014°46'E) is :

- **a** 1000ft
- **b** FL60
- c FL100
- **d** 2500 ft

33.04.02.05. Standards Instrument Departures

219 Unless otherwise shown on charts for standard instrument departure the routes are given with:

- a true course
- **b** magnetic headings
- c magnetic course
- **d** true headings

(For this question use Route Manual chart SID PARIS Charles-De-Gaulle (20-3))
 ²³²⁷ Planning a IFR flight from Paris (Charles de Gaulle) to London (Heathrow). Find the elevation of the departure aerodrome.

- **a** 217 ft.
- **b** 268 ft.
- c 387 ft.

d 2 ft.

221		(For this question use Route Manual chart STAR LONDON Heathrow (10-2))
id	2328	Planning a IFR flight from Paris (Charles de Gaulle) to London (Heathrow). Find the
		elevation of the destination aerodrome.

- a 77 ft.
- b 80 ft.
- **c** 177 ft.
- **d** 100 ft.

	222	Planning an IFR-flight from Paris to London (Heathrow). Name the identifier and
id	4390	frequency of the initial approach fix (IAF) of the BIG 2A arrival route.
а	EPM	316 kHz
b	BIG 1	15.1 MHz
С	OCK	15.3 MHz
d	BIG 1	15.1 kHz
2	223	The minimum holding altitude (MHA) and maximum holding speed (IAS) at MHA at
Id	4409	OCKHAM OCK 115.3 are:
а	7000	t and 220kt
b	9000f	and 220kt
C	7000f	and 250kt
d	9000f	and 250kt
2 id	2 24	The route distance from CHIEVRES (CIV) to BOURSONNE (BSN) is :
а	88 NIM	
u h	83 NM	
c c	96 NM	
d	73 NM	
	225	ECUL: Which of the following is a correct Minimum Sofe Altitude (MSA) for the
⊿ id	4411	Airport?
а	West	sector 2300 ft within 25 NM
b	West	sector 2100 ft within 25 NM
С	East s	ector 2100 ft within 50 NM
d	East s	ector 2300 ft within 50 NM
2	226	LEMD: For runway 33 arrivals from the east and south, the Initial Approach Fix
id	4412	(IAF) inbound from airway UR10 is :
а		
	CJIN	
b	VTB	
b c	VTB CENT	٩
b c d	VTB CENT MOTII	A
b c d 2	VTB CENT MOTI	A
b c d 2 id	VTB CENT MOTII 227 4413	A
b c d 2 id a	VTB CENT MOTII 227 4413 ALBIX	A
b c d 2 id a b	VTB CENT MOTII 227 4413 ALBIX ALBIX	A
b d ² ^{id} a b c	VTB CENT MOTII 227 4413 ALBIX ALBIX	A
b c d 2 id a b c d	VTB CENT MOTII 227 4413 ALBIX ALBIX ALBIX	A
b c d ² id a b c d ² id a b c d ²	VTB CENT MOTII 227 4413 ALBIX ALBIX ALBIX 228	A LSZH: Which is the correct ALBIX departure via AARAU for runway 16? 7A (7S 6H 6E EHAM: The route distance from runway 27 to ARNEM is:
b c d 2 id a b c d 2 id a b c d 2 id a b c d 2 id a b c d 2 id a b c d 2 id a b c d	VTB CENT MOTII 227 4413 ALBIX ALBIX ALBIX 228 4414 52 NM	A LSZH: Which is the correct ALBIX departure via AARAU for runway 16? 7A X 7S 6H 6E EHAM: The route distance from runway 27 to ARNEM is:
b c d 2 id a b c d 2 id a b	VTB CENT MOTII 227 4413 ALBIX ALBIX ALBIX ALBIX 228 4414 52 NM 35 NM	A LSZH: Which is the correct ALBIX departure via AARAU for runway 16? 7A (7S 6H 6E EHAM: The route distance from runway 27 to ARNEM is:
b d 2 id a b c d 2 id a b c d 2 id a b c d 2 id a b c c d	VTB CENT MOTI 227 4413 ALBIX ALBIX ALBIX 228 4414 52 NM 35 NM	A LSZH: Which is the correct ALBIX departure via AARAU for runway 16? 7A (7S 6H 6E EHAM: The route distance from runway 27 to ARNEM is:

229 EHAM: Which of the following statements is correct for ANDIK departures from

- id 4415 runway 19L?
- a Maximum IAS 250kt turning left at SPL 3.1 DME
- **b** Cross ANDIK below FL60
- c The distance to ANDIK is 25 NM

d Contact SCHIPOL DEPARTURE 119.05 passing 2000 ft and report altitude

- **230** EDDM: The correct arrival route and Initial Approach Fix (IAF) for an arrival from the west via TANGO for runway 08 L/R is:
- a AALEN 1T, IAF MBG

b AALEN 1T, IAF ROKIL

- c NDG 1T, IAF ROKIL
- d DKB 1T, IAF ROKIL

231 LSZH: Aeroplane arriving via route BLM 2Z only, should follow the following route to EKRON int:

- a WIL R018 outbound to EKRON int
- b TRA R247 outbound to EKRON int

c BLM R111 to GOLKE int then TRA R-247 inbound to EKRON int

d HOC R067 via GOLKE to EKRON int

232 EDDM: Which is the correct departure via KEMPTEN from runway 26L?

id 5101

a KEMPTEN THREE ECHO

b KEMPTEN FIVE SIERRA

c KEMPTEN THREE QUEBEC

d KEMPTEN THREE NOVEMBER

33.04.02.06. MNPS-RVSM

233Your aircraft is approved for MNPS and RVSM. What do you have to insert in itemid896110 of the ATC flight plan?

- a W
- bΥ
- сΧ

d W, X

234 Which is the correct date of the implementation of the RVSM in the European Airspace?

- a 1 January 2002
- b 24 January 2002
- c 1 February 2005
- **d** 1 July 2001

235 RVSM In-Flight procedure: Cross checks of the primary altimeters shall be made at intervals of approximately one hour. These primary altimeters shall agree within

- **a** 50 ft
- **b** 100 ft
- **c** 150 ft
- d 200 ft

2	236	RVSM Pre-Flight procedure: The flight crew shall verify the altimetry accuracy by
id	8964	setting the QNH or QFE. The reading should then agree with the altitude of the
		apron or the zero height indication within
а	25 ft	
b	30 ft	
С	75 ft	
d	150 ft	
2	237	Which flight level is not a RVSM level?
id	8965	
а	FL 28	0
b	FL 29)
С	FL 30)
d	FL 31)
2	238	Which equipment failure must not be reported to ATC on a RVSM level?
id	8966	
а	Loss of	of thrust on one or more engines which requires a descent
b	Main	hydraulic-pump failure
С	Loss	of one or more altimetry systems
d	Failur	e of all automatic altitude-control systems
2	239	Do you need TCAS/ACAS Version 7.0 to operate in EUR-RVSM airspace?
id	8967	f the circult has more than 20 secto or the circult weight is even 15000kg
a	res, i	the aircraft has more than 30 seats or the aircraft weight is over 15000kg
D		
C		,
a		
2	240	Your aircraft is not RVSM approved. Are you able to enter RVSM airspace?
iu a	8908 Vos I	nut not as a civil operator
a h	Voc k	
с С	Ves h	nut I can climb and descend through RVSM airspace only
ч	Ves o	anly on El 310 and El 350
<u> </u>	100, 0	
∠ id	2 4 8969	
а	Horizo	ontal measuring unit
b	Heigh	t measuring unit
с	Heigh	t metering unit
d	Heigh	t monitoring unit
2	42	RVSM In-Flight procedure: When changing levels, the aircraft shall not overshoot
id	8970	or undershoot the cleared flight level by more than
а	50 ft	

- b 150 ft
- **c** 200 ft
- **d** 100 ft

243 What is the proper phraseology if ATC wants to know if you are RVSM approved? id 8971

- a RVSM ok
- **b** RVSM approved
- c Affirmative RVSM
- d Affirm RVSM

244 Which equipment is not necessary to get a RVSM approval?

- **id** 8972
- **a** Altitude alerting system
- **b** GPS with altitude reporting system
- ${\boldsymbol{c}}$ Automatic altitude control system
- d SSR transponder with altitude reporting system in use for altitude keeping

2 id	2 45 8973	When approaching a cleared Flight level, the vertical speed should not exceed
а	2000	ft/min.
b	350 ft	/min.
С	750 ft	/min.
d	1500	ft/min.
2	246	Which document provides guidance for the approval of RVSM aircraft?
id	8974	
а	AIC 2	2
b	JAR-F	-CL
С	JAA .	ΓGL No. 6
d	IL 20	
2	247	RVSM was first implemented in which airspace?
id	8975	
а	Pacifi	C
b	Europ	e
С	NAT	
d	Africa	
2	18	
id	0	Which transponder code is correct if you are 40 minutes before entering NAT
	8976	Which transponder code is correct if you are 40 minutes before entering NAT airspace?
а	8976 7500	Which transponder code is correct if you are 40 minutes before entering NAT airspace?
a b	8976 7500 2000	Which transponder code is correct if you are 40 minutes before entering NAT airspace?
a b c	8976 7500 2000 as re	Which transponder code is correct if you are 40 minutes before entering NAT airspace?
a b c d	8976 7500 2000 as re 7000	Which transponder code is correct if you are 40 minutes before entering NAT airspace?
a b c d	8976 7500 2000 as re 7000	Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you
a b c d 2 id	8976 7500 2000 as re 7000 249 8977	Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track?
a b d d id a	 8976 7500 2000 as reg 7000 249 8977 2 NIM 	Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track?
a b d 2 id a b	 8976 7500 2000 as re 7000 249 8977 2 NM 2 NM 	Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track?
a b d 2 id a b c	 2000 2000<th>Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track? left <i>I</i> left or right</th>	Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track? left <i>I</i> left or right
a b c d 2 id a b c d	 2000 2000<th>Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track? left <i>I</i> left or right A left</th>	Which transponder code is correct if you are 40 minutes before entering NAT airspace? quested by ATC In case of an engine failure, unable to maintain altitude, how many miles do you have to fly offset of NAT track? left <i>I</i> left or right A left

- **250** What is the Polar Track System?
- **id** 8978
- a 5 routes between Alaska over the North Pole to Japan
- **b** 2 flex tracks from Tokyo to Honolulu
- c 6 tracks between Hawaii and USA

d Fixed tracks between Europe over the North Pole to Alaska

251 You are entering the NAT. What is the tolerance of the boundary window?

- id 8979
- a 5 minutes
- **b** 3 minutes, but only between FL 310 and FL 390
- c 3 minutes
- ${\boldsymbol{\mathsf{d}}}$ no boundary window required
- 252 What is the NAT Track Message?

id 8980

- **a** The publication of the Preferred Route Message (PRM)
- **b** The ATC clearance given before the boundary window
- c The complete MNPS flight plan

d The publication of the Organized Track Message (OTS)

- 253 You are flying from ZRH to JFK (EET 8h04'). The EOBT is 1800Z. Are you able to id 8981 fly the following track? A 54/15 55/20 55/30 53/40 51/50 CYMON EAST LVLS NIL WEST LVLS 310 330 340 350 360 370 390 EUR RTS VIA BABAN NAR N144B N148B-
- a Yes
- b No
- c Yes, but only via exit point BABAN

d Yes, but only via exit point CYMON

- 254 Your position is N50° W20°. Your altimeter shows FL263 descending. Which
- id 8982 airspace is that?
- аA
- **b** MNPS
- c F
- d C

255 Your HF radio is u/s. Are you able to cross the NAT?

id 8983

- a No, before departure, HF radio is a must
- b Yes, I can fly the OTS
- c Yes, but only on special routes
- d It is up to you

- id 8984 What is your action in case you have one system left?
- **a** The Copilot has to exchange the two black boxes
- b Continue, because only one system must be operative
- c Fly 15NM offset to the normal tracks
- d File Special routes or fly above or below the MNPS

257 Day time OTS are valid between...

- **id** 8985
- a 1130 LT until 1800 LT
- **b** 1130 Z until 1800 LT
- c 1130 Z until 1800 Z
- **d** 0100 Z until 0900 Z

258 On which VHF frequency can you obtain the NAT clearance from Shanwick?

a 127. 65 (if your aircraft is registered E of 30 W)

- **b** 120.00 (if your aircraft is registered W of 30 W)
- c 123.95 (if your aircraft is registered E of 30 W)
- d only possible on HF

33.04.02.07. ETOPS

- **259** An aerodrome with weather reports indicating that the weather conditions are at or above operation minima from one hour prior to one hour after the anticipated arrival is defined as:
- a adequate
- b suitable
- c not suitable and not adequate
- d enroute alternate

2	260	On the ground in ZRH the APU on your B737-300 cannot be started. Can you
id	8997	accept the aircraft for an ETOPS flight?
а	Yes, o	company procedures do not required it's use
h	No	

- c Yes, providing both engine driven generators operates normally
- d Yes, as it is not a required item to dispatch the aircraft
- 261 What is the Extended Range Entry point (or ETOPS entry point)?
- **id** 8998
- a The ETP
- **b** The point on the route which is 120 minutes flying time (with approved single engine cruise speed) from a suitable alternate
- c The point of the route which is 60 minutes flying time (with approved single engine cruise speed) from an alternate airport
- d The point where you enter the extended range speed into the FMC
- 262 What is general the "most critical fuel scenario" on the B737-300?

id 8999

- a The two engine fuel scenario
- **b** Drift down without APU to 10'000 ft
- c One engine fuel scenario
- d If you have less fuel than for 30 minutes on board

2 id	2 63 9000	(North Atlantic Plotting Chart, ETOPS) You are flying from Shannon (EINN) to Keflavik (BIKF). The wind component to BIKF is 10kts headwind and to Shannon you will have 20kts tailwind. The ETP from EINN to BIKF is?
а	416 N	IM from EINN
b	400 N	IM from BIKF
С	384 N	IM from EINN
d	416 N	IM from BIKF
2 id	2 64 9001	North Atlantic Plotting Chart, ETOPS) You are flying from Santa Maria (LPLA) to St. John's (CYYT). The wind component to CYYT is 30kts headwind and to LPLA you will have 20kts tailwind. The ETP from SMA to YYT is?
а	580 N	IM from YYT
b	580 N	IM from YQX
С	580 N	IM from SMA
d	650 N	IM from SMA
2 id	2 65 9002	(Critical fuel reserves long range cruise, CAP698) You have an engine failure and a decompression at the same time. Your data are: Tailwind: 25 kts Distance to diversion airport: 820 kts ISA: +10°C Weight: 55'000 kg lcing conditions: YES What is your diversion fuel?
а	8300	kg
b	7035	kg
С	7000	kg
d	8440	kg
2 id	2 66 9003	(Critical fuel reserves long range cruise, CAP698) You have a decompression at your cruising altitude and following information: Tailwind: 25 kts Distance to diversion airport: 820 kts ISA: +20°C Weight: 55'000 kg lcing conditions: No What is your diversion fuel?
а	7270	kg
b	7000	kg
С	8581	kg
d	7480	kg
2	267	(Area of Operation, CAP 698) You have following information: Weight: 57.5 t
id	9004	Speed schedule: LRC ETOPS approval: 180 min What is your area of operation?
а	1169	NM
b	804 N	M
С	1100	NM
d	1134	NM
2 id	2 68 9005	Who is able to perform an ETOPS pre-departure service check on an A330?
а	Every	mechanic who has an A330 licence
b	Only	an ETOPS qualified maintenance person can do that
С	Only [·]	TMC is allowed to do that
d	No pr	e-departue service checks are required

33.04.03. General flight planning tasks

33.04.03.01. Checking of AIP and NOTAM

0	0.01.	
2 id	2 69 5077	From which of the following would you expect to find information regarding known short unserviceability of VOR, TACAN, and NDB ?
а	NOTA	M
b	AIP (A	Air Information Publication)
С	SIGM	ET
d	ATCC	broadcasts
2 id	2 70 5078	From which of the following would you expect to find the dates and times when temporary danger areas are active
а	Only /	AIP (Air Information Publication)
b	NOTA	AM and AIP (Air Information Publication)
С	SIGM	ET
d	RAD/I	NAV charts
2 id	2 71 5079	From which of the following would you expect to find details of the Search and Rescue organisation and procedures (SAR) ?
а	ATCC	broadcasts
b	AIP (/	Air Information Publication)
С	NOTA	M
d	SIGM	ET
2 id	2 72 5080	From which of the following would you expect to find facilitation information (FAL) regarding customs and health formalities ?
а	NOTA	M
b	NAV/	RAD charts
С	ATCC	
d	AIP (/	Air Information Publication)
33	3.04.	03.02. Selection of altitudes or flight levels
2 id	2 73 2051	(For this question use Route Manual chart E(HI)4) An aeroplane has to fly from Abbeville (50°08.1'N 001°51.3'E) to Biggin (51°19.8'N 00°00.2'E). What is the first FL above FL295 that can be flown on an IFR flightplan ?

- **a** FL 330
- b FL 310
- **c** FL 320
- **d** FL 300
- 274(For this Question use Fuel Planning MRJT1 Fig. 4.2.1) Find the OPTIMUMid43764376ALTITUDE for the twin jet aeroplane. Given: Cruise mass=54000 kg, Long range
cruise or .74 MACH
- a 33800 ft
- b 34500 ft
- **c** 35300 ft
- d maximum operating altitude

2 id	275	(For this Question use Fuel Planning MRJT1 Fig. 4.2.1) Find the OPTIMUM
	4377	ALTITODE for the twin jet aeropiane. Given, Cruise mass=50000 kg, .78 MACH
a h	30200	ft
0	36700	ft
d d	maxim	num operating altitude
<u> </u>	74	(For this Question use Eucl Planning MP IT1) Find the ELIEL MILEAGE PENALTY
⊿ id	4378	for the twin iet aeroplane with regard to the given FLIGHT LEVEL. Given: Long
	I	range cruise, Cruise mass=53000 kg, FL 310
а	4 %	
b	1 %	
С	10 %	
d	0 %	
2	.77	An appropriate flight level for flight on airway UR1 from ORTAC (50°00'N
	5090	002*00 W) to MIDHORST MID 114.0 (51*03 N 000*37 W) IS:
a h		
C C	FL200	
о d	FL250	
	78	An appropriate flight level for flight on airway LIG1 from ERI ANGEN ERI, 114.9
id	5091	(49°39°'N 011°09'E) to FRANKFURT FFM 114.2 (50°03'N 008°38'E) is :
а	FL310	
b	FL290	
С	FL300	
d	FL320	
2	.79	An appropriate flight level for flight on airway UG5 from MENDE-NASBINALS MEN
id	5093	115.3 (44°36'N 003°10'E) to GAILLAC GAI 115.8 (43°57'N 001°50'E) is :
а	FL300	
b	FL280	
C	FL290	
a	FL310	An engranded flight level for flight on einvers LID04 from NANTEO NTO 447.0
id	5094	An appropriate high level of high on allway $OR24$ from NANTES NTS T17.2 (47°09'N 001°37'W) to CAEN CAN 115.4 (49°10'N 000°27'W) is
а	FL 300	
b	FL290	
с	FL310	
d	FL270	
2	81	An appropriate flight level for flight on airway B3 from CHATILLON CTL 117.6
id	5095	(49°08'N 003°35'E) to CAMBRAI CMB 112.6 (50°14'N 003°09'E) is :
а	FL80	
b	FL170	
С	FL60	
d	FL50	

2	2 82	An appropriate flight level for flight on airway R10 from MONTMEDY MMD 109.4
ю 2		(49 24 N 005 08 E) 10 CHATTELON CTE TT7.0 (49 08 N 005 55 E) 15.
a h	FL/0	
с С	FL 50	
0 d	FI 40	
	002	An appropriate flight level for IER flight in accordance with semi-circular height
id	5098	rules on a course of 180° (M) is:
а	FL100	
b	FL90	
С	FL95	
d	FL105	
2	284	An appropriate flight level for IFR flight in accordance with semi-circular height
id	5099	rules on a magnetic course of 200° is:
а	FL320	
b	FL310	
С	FL290	
d	FL300	
2 id	2 85 5534	You must fly IFR on an airway orientated 135° magnetic with a MSA at 7 800 ft. Knowing the ONH is 1 025 bPa and the temperature is $ISA + 10^{\circ}$ the minimum
	5554	flight level you must fly at is:
а	75	
b	80	
с	90	
d	70	
2	286	An aircraft, following a 215° true track, must fly over a 10 600 ft obstacle with a
Ia	5535	airport close by which is almost at sea-level, is 1035 and the temperature is ISA -
		15°C, the minimum flight level will be:
а	140	
b	120	
С	130	
d	150	
2	287	On an IFR navigation chart, in a 1° quadrant of longitude and latitude, appears the
id	5538	following information "80". This means that within this quadrant:
a L	the m	Inimum safe altitude is 8 000 ft
D	the of	himum llight level is FL 80
с С	the flo	or of the airway is at 8,000 ft
		On an instrument approach short, a minimum sector altitude (MSA) is defined in
⊿ id	5546	relation to a radio navigation facility. Without any particular specification on distance, this altitude is valid to:
а	20 NIV	
b	25 NN	
С	15 NIV	
d	10 NIV	
33.	04. IFR	(AIRWAYS) FLIGHT PLANNING 2002/12/22 Page 48 of 66

289 An IFR flight is planned outside airways on a course of 235° magnetic. The
 ^{id} 5547 minimum safe altitude is 7800 ft. Knowing the QNH is 995 hPa, the minimum flight level you must fly is:

- **a** 80
- **b** 90
- **c** 85

d 100

33.04.03.05. Completion of fuel plan

- An aeroplane has the following masses: ESTLWT= 50 000 kg Trip fuel= 4 300 kg
 ¹³⁹⁴ Contingency fuel= 215 kg Alternate fuel (final reserve included)= 2 100kg Taxi= 500 kg Block fuel= 7 115 kg Before departure the captain orders to make the block fuel
 9 000 kg. The trip fuel in the operational flight plan should read:
- a 4 300 kg.
- **b** 6 185 kg.
- **c** 9 000 kg.
- **d** 6 400 kg.

33.04.03.06. Preliminary study of instrument approach

291 EGLL ILS DME Rwy 09L: The Decision Altitude (DA) for a ILS straight-in landing is : id 4028

- a 280 ft
- **b** 200 ft
- **c** 400 ft
- **d** 480 ft

292 EHAM VORDME Rwy 22: The Missed Approach procedure is to climb to an alitude of (i)------ on a track of (ii) -----

- a (i) 3000 ft (ii) 223°
- **b** (i) 200 ft (ii) 223°
- **c** (i) 3000 ft (ii) 160°
- d (i) 2000 ft (ii) 160°

293 LSZH ILS Rwy 14: The minimum glide slope interception altitude for a full ILS is:

- **id** 5103
- a 4000 ft
- **b** 3370 ft
- **c** 2598 ft
- **d** 1968 ft

294 The Radio Altimeter minimum altitude for a CAT 2 ILS DME :

- **id** 5104
- a 100 ft
- **b** 88 ft
- **c** 300 ft
- **d** 188 ft

2 id	2 95	EGLL ILS DME Rwy 09R: The Minimum Descent Altitude (MDA) for an ILS glide
а	275 ft	
b	405 ft	
c	480 ft	
d	200 ft	
-2	96	LEPG. VORDME Rwy 27. The crossing altitude and descent instruction for a
id	5106	propeller aircraft at COULOMMIERS (CLM) are :
а	Cross	at FL60 and maintain
b	Cross	at FL70 descend to 4000 ft
С	Cross	at FL80 descend to FL70
d	Cross	at FL60 descend to 4000 ft
2	297	EDDM ILS Rwy 26R: The ILS frequency and identifier are:
id	5107	
а	108.7	IMSW
b	108.7	IMNW
С	108.3	IMNW
d	108.3	IMSW
2	2 98	LFPG ILS Rwy 09: The ILS localizer course is :
ю э	100°	
a h	000 088°	
с С	118°	
d	268°	
		EDDM NDD DME Dury 26Ly The frequency and identifier of the NDD for the
⊿ id	5110	published approachs are:
а	ا 108.6	DMS
b	338 M	NW
с	400 M	SW
d	112.3	MUN
3	800	Which approach segment starts at the point were you report "established" ?
id	8857	
а	Final a	approach
b	Initial	approach
С	Go ar	bund
d	Intern	nediate approach
3	801	Which approach segment starts at the FAF and ends at the MAP?
id	8858	
a	Initial	approach
b	Final	approach
C	Go ar	bund
d	Interm	ediate approach

33.04.04. IR Flightplan exercices

302 Planning an IFR-flight from Paris to London for the twin jet aeroplane. Given:

id 4564 Estimated Landing Mass 49700 kg, FL 280, W/V 280°/40 kt, Average True Course 320°, Procedure for descent .74 M/250 KIAS Determine the distance from the top of descent to London (elevation 80 ft).

- **a** 87 NM
- b 76 NM
- **c** 97 NM
- **d** 65 NM

303 Planning an IFR-flight from Paris to London for the twin jet aeroplane. Given:

- id 4565 Estimated Landing Mass 49700 kg, FL 280, W/V 280°/40 kt, Average True Course 320°, Procedure for descent .74 M/250 KIAS Determine the time from the top of descent to London (elevation 80 ft).
- **a** 10 min
- b 19 min
- **c** 17 min
- **d** 8 min
- 304 Planning an IFR-flight from Paris to London for the twin jet aeroplane. Given:
 ^{id} 4566 Estimated Landing Mass 49700 kg, FL 280, W/V 280°/40 kt, Average True Course 320°, Procedure for descent .74 M/250 KIAS Determine the fuel consumption from the top of descent to London (elevation 80 ft).
- a 263 kg
- b 273 kg
- **c** 210 kg
- **d** 320 kg

33.05. JET AEROPLANES FLIGHT PLANNING

33.05.01. Additional flight planning aspects for jet aeropla 33.05.01.01. Fuel planning

305 Mark the correct statement: If a decision point procedure is applied for flight planning,

 ${\boldsymbol{a}}$ the trip fuel to the destination aerodrome is to be calculated via the suitable enroute alternate.

b the trip fuel to the destination aerodrome is to be calculated via the decision point.

c a destination alternate is not required.

 \mathbf{d} the fuel calculation is based on a contingency fuel from departure aerodrome to the decision point.

306 An operator (turbojet engine) shall ensure that calculation up of usable fuel for a flight for which no destination alternate is required includes, taxi fuel, trip fuel, contingency fuel and fuel to fly for:

- **a** 45 minutes plus 15% of the flight time planned to be spent at cruising level or two hours whichever is less
- **b** 2 hours at normal cruise consumption

c 30 minutes at holding speed at 450 m above aerodrome elevation in standard conditions

 ${\rm d}\,$ 30 minutes at holding speed at 450 m above MSL in standard conditions

307 Planning a flight from Paris (Charles de Gaulle) to London (Heathrow) for a twin -

^{id} 1783 jet aeroplane. Preplanning: Maximum Take-off Mass: 62 800 kg Maximum Zero
 Fuel Mass: 51 250 kg Maximum Landing Mass: 54 900 kg Maximum Taxi Mass: 63
 050 kg Assume the following preplanning results: Trip fuel: 1 800 kg Alternate fuel:
 1 400 kg Holding fuel (final reserve): 1 225 k

- **a** 55 765 kg.
- b 51 515 kg.
- **c** 51 425 kg.
- **d** 52 265 kg.

308 The required time for final reserve fuel for turbojet aeroplane is:

- **id** 1855
- **a** 45 min.
- b 30 min.
- **c** 60 min.
- d Variable with wind velocity.

309 The quantity of fuel which is calculated to be necessary for a jet aeroplane to fly ¹⁹⁷⁸ IFR from departure aerodrome to the destination aerodrome is 5352 kg. Fuel consumption in holding mode is 6 000 kg/h. Alternate fuel is 4380 kg. Contingency should be 5% of trip fuel. What is the minimum required quantity of fuel which should be on board at take-off?

- **a** 13370 kg.
- **b** 14500 kg.
- c 13000 kg.
- d 13220 kg.

3 id	10 1979	The following fuel consumption figures are given for a jet aeroplane: -standard taxi fuel: 600 kgaverage cruise consumption: 10 000 kg/hholding fuel consumption at 1500 ft above alternate airfield elevation: 8000 kg/hflight time from departure to
		destination: 6 hours -fuel for diversion to alternate: 10 200 kg. The minimum ramp fuel load is:
а	77 80	D ka
b	74 80	0 kg
c	79 80	0 kg
ď	77 20) ka
	11	A jet aeroplane has a cruising fuel consumption of 4060 kg/h, and 3690 kg/h during
id	1981	holding. If the destination is an isolated airfield, the aeroplane must carry, in addition to contingency reserves, additionnal fuel of :
а	7380 I	kg.
b	8120	kg.
С	1845 I	<g.< th=""></g.<>
d	3500 I	<g.< th=""></g.<>
3	12	A jet aeroplane is to fly from A to B. The minimum final reserve fuel must allow for :
id	1982	
а	20 mii	nutes hold over alternate airfield.
b	30 mi requi	nutes hold at 1500 ft above destination aerodrome elevation, when no alternate is red.
С	30 mir	nutes hold at 1500 ft above mean sea level.
d	15 miı	nutes hold at 1500 ft above destination aerodrome elevation.
3	13	(For this question use Flight Planning Manual MRJT 1 Figure 4.3.1.B) Given :
id	2060	estimated zero fuel mass 50 t; estimated landing mass at alternate 52 t; final reserve fuel 2 t; alternate fuel 1 t; flight to destination, distance 720 NM, true course
		(TC) 030, W/V 340/30; cruise: long range FL 330, outside air temperature -30 ° C.
2	1 800	
u h	4 400	kg; 01 : 45
0	4 750	kg; 02 : 00
с А	4 600	kg; 02 : 05
_u	4 000	(For this question use Elight Dianning Manual MP IT 1 Figure 4.2.6) Civen:
ۍ id	2061	estimated dry operation mass 35 500 kg; estimated load 14 500 kg; final reserve
		component 10 kt Find : fuel and time to alternate.
а	1 100	ka 44 min
∽ h	1 100	ka: 25 min
c	800 k	n: 24 min
d	800 k	r = 40 min
		(For this question use Elight Planning Manual MP IT 1 Figure 4.3.3C) Given: ground
id	2062	distance to destination aerodrome 1 600 NM; headwind component 50 kt; FL 330; cruise 0.78 Mach; ISA + 20 ° C; estimated landing weight 55000 kg . Find: simplified flight planning to determine estimated trip fuel and trip time.
а	12 40	0 kg. 04h 12 min
b	11 40	0 kg. 04h 12 min
С	12 40	0 kg. 03h 55 min
d	11 40	0 kg. 03h 55 min

316 (For this question use Flight Planning Manual MRJT 1 Figure 4.4) Given: dry ^{id} ²⁰⁶³ operating mass 35 500 kg; estimated load 12 000 kg, contingency approach and landing fuel 2 500 kg; elevation at departure aerodrome 500 ft; elevation at alternate
aerodrome 30 ft. Find: final reserve fuel for a jet aeroplane (holding) and give the elevation which is relevant.
a 2 360 kg;destination elevation
b 2 360 kg; alternate elevation
c 1 180 kg;destination elevation
d 1 180 kg; alternate elevation
317 The purpose of the decision point procedure is ?
a To increase the safety of the flight.
b To reduce the landing weight and thus reduce the structural stress on the aircraft.
c To reduce the minimum required fuel and therefore be able to increase the traffic load.
d To increase the amount of extra fuel.
318 When using decision point procedure, you reduce the
d 2186
a moluling rule by 30%.
destination.
c reserve fuel from 10% down to 5%.
d contingency fuel by adding contingency only from the burnoff between decision point and destination.
319 (For this Question use Fuel Planning MRJT1) Find the SPECIFIC RANGE for the twin jet aeroplane flying below the optimum altitude (range loss = 6%) and using the following data. Given: MACH .74 CRUISE, Flight level = 310, Gross mass = 50000 kg, ISA conditions
a 2807 NAM/1000 kg
b 187 NAM/1000 kg
c 2994 NAM/1000 kg
d 176 NAM/1000 kg
320 (For this Question use Fuel Planning MRJT1) Find the FUEL FLOW for the twin jet id 2766 aeroplane with regard to the following data. Given: MACH .74 cruise, Flight level 310, Gross mass 50000 kg, ISA conditions
a 1497 kg/h
b 1150 kg/h
c 2300 kg/h
d 2994 kg/h
 321 (For this Question use Fuel Planning MRJT1 Fig. 4.3.6) In order to find id 4371 ALTERNATE FUEL and TIME TO ALTERNATE, the AEROPLANE OPERATING MANUAL shall be entered with:
a distance in nautical miles (NM), wind component, landing mass at alternate
b distance in nautical air miles (NAM), wind component, landing mass at alternate
c distance in nautical miles (NM), wind component, zero fuel mass

d distance in nautical miles (NM), wind component, dry operating mass plus holding fuel

_			
3 id	322 4372	The final reserve fuel for aeroplanes with turbine engines is	
а	fuel to condit	fly for 45 minutes at holding speed at 1000 ft (300 m) above aerodrome elevation in standard ions.	
b	fuel to fly for 45 minutes at holding speed at 1500 ft (450 m) above aerodrome elevation in standard conditions.		
С	fuel to fly for 30 minutes at holding speed at 1500 ft (450 m) above aerodrome elevation in standard conditions.		
d	fuel to condit	fly for 60 minutes at holding speed at 1500 ft (450 m) above aerodrome elevation in standard ions.	
3 id	323 4373	Which of the following statements is relevant for forming route portions in integrated range flight planning?	
а	The d	istance from take-off up to the top of climb has to be known.	
b	No se	gment shall be more than 30 minutes of flight time.	
С	Each	reporting point requires a new segment.	
d	A sma	all change of temperature (2 °C) can divide a segment.	
3 id	324 4379	(For this Question use Fuel Planning MRJT1) Find: Final fuel consumption for this leg Given: Long range cruise, Temperature -63°C, FL 330, Initial gross mass enroute 54100 kg, Leg flight time 29 min	
а	1100	kg	
b	1107	kg	
С	1093	kg	
d	1000	kg	
3 id	4380	(For this Question use Fuel Planning MRJT1) Find: Air distance in Nautical Air Miles (NAM) for this leg and fuel consumption Given: Flight time from top of climb at FL 280 to the enroute point is 48 minutes. Cruise procedure is long range cruise. Temperature is ISA -5°C. The take-off mass is 56000 kg and climb fuel 1100 kg.	
а	345 N	AM; 2000 kg	
b	349 N	AM; 2000 kg	
С	345 N	AM; 1994 kg	
d	345 N	AM; 2006 kg	
3 id	326 4381	(For this Question use Fuel Planning MRJT1) Given: Brake release mass 57500 kg, Initial FL 280, average temperature during climb ISA -10°C, average head wind component 18 kt Find: Climb time for enroute climb 280/.74	
а	13 mi	n	
b	11 mir	1	
С	15 mir	1	
d	14 mir	۱ 	
3 id	327 4382	(For this Question use Fuel Planning MRJT1) Given: Brake release mass 57500 kg, Temperature ISA -10°C, Headwind component 16 kt, Initial FL 280 Find: Still air distance (NAM) and ground distance (NM) for the enroute climb 280/.74	
а	62 NA	M, 59 NM	
b	59 NA	M, 62 NM	
С	62 NA	M, 71 NM	
d	71 NA	M, 67 NM	

- **328** (For this Question use Fuel Planning MRJT1) Given: Brake release mass 57500 kg, Temperature ISA -10°C, Average headwind component 16 kt, Initial FL 280 Find: Climb fuel for enroute climb 280/.74
- **a** 1040 kg
- **b** 1238 kg
- **c** 1387 kg
- d 1138 kg
- **329** (For this Question use Fuel Planning MRJT1) Given: Long range cruise, OAT -^{id} 4386 45°C at FL 350, Gross mass at the beginning of the leg 40000 kg, Gross mass at the end of the leg 39000 kg Find: True air speed (TAS) and cruise distance (NAM) for a twin jet aeroplane
- a TAS 423 kt, 227 NAM
- b TAS 433 kt, 227 NAM
- **c** TAS 433 kt, 1163 NAM
- d TAS 423 kt, 936 NAM
- 330 (For this Question use Fuel Planning MRJT1) Given: Estimated take-off mass
 ⁴³⁸⁷ 57000 kg, Ground distance 150 NM, Temperature ISA -10°C, Cruise at .74 Mach
 Find: Cruise altitude and expected true air speed
- a 25000 ft, 435 kt
- **b** 24000 ft, 445 kt
- **c** 33500 ft, 430 kt
- d 33900 ft, 420 kt
- **331** (For this Question use Fuel Planning MRJT1) Given: twin jet aeroplane, FL 330, ^{id} ⁴⁵⁵¹ Long range cruise, Outside air temperature -63°C, Gross mass 50500 kg Find: True air speed (TAS)
- **a** 433 kt
- b 420 kt
- **c** 431 kt
- **d** 418 kt

332 (For this Question use Fuel Planning MRJT1) Given: Diversion distance 720NM
 ^{id} 5730 Tail wind component 25kt Mass at point of diversion 55000kg Temperature ISA Diversion fuel available 4250kg What is the minimum pressure altitude at which the above conditions may be met ?

- **a** 26000ft
- b 20000ft
- **c** 16000ft
- **d** 14500ft

333 (For this Question use Fuel Planning MRJT1) Given: Diversion distance 650 NM ^{id} 5731 Diversion pressure altitude 16 000 ft Mass at point of diversion 57 000 kg Head wind component 20 kt Temperature ISA + 15°C The diversion (a) fuel required and (b) time, are approximately :

- a (a) 4400kg (b) 1h 35min
- **b** (a) 3900kg (b) 1h 45min
- **c** (a) 6200kg (b) 2h 10min
- d (a) 4800kg (b) 2h 03min

3 id	334 5732	(For this Question use Fuel Planning MRJT1) Given: Distance to alternate 950
iu.	5752	Diversion fuel available 5800kg The minimum pressure altitude at which the
		above conditions may be met is :
а	22000	ft
b	20000	ft
с	26000	ft
d	18000	ft
3	335	(For this Question use Fuel Planning MRJT1) A descent is planned at .74/250KIAS
id	5736	from 35000ft to 5000ft. How much fuel will be consumed during this descent?
а	290kg	I
b	150kg	
С	278kg	I
d	140kg	
33	3.05.	01.02. Computation of critical point (CP)
3	336	If CAS is 190 kts, Altitude 9000 ft. Temp. ISA - 10°C, True Course (TC) 350°, W/V
id	2193	320/40, distance from departure to destination is 350 NM, endurance 3 hours, and
		actual time of departure is 1105 UTC. The Point of Equal Time (PET) is reached at
_	4040	
a	1213	
D	1221 (
C	1233 (
a	1203 (
id	2194	If CAS is 190 kts, Altitude 9000 ft. Temp. ISA - 10°C, True Course (TC) 350°, W/V 320/40, distance from departure to destination is 350 NM, endurance 3 hours and
		actual time of departure is 1105 UTC. The distance from departure to Point of
		Equal Time (PET) is :
а	183 N	M
b	147 N	M
С	203 N	M
d	167 N	M
3	338	Find the distance from waypoint 3 (WP 3) to the critical point. Given: distance from
id	2772	WP 3 to WP 4 = 750 NM, TAS out 430 kt, TAS return 425 kt, Tailwind component
_	400 N	
a L	406 N	
D	OTE N	
	375 N	M
۔ C	375 N 342 N	M M
c d	375 N 342 N 403 N	M M Find the time to the Point of Safe Poture (PSP). Civer: Maximum useship fuel
c d 3 id	375 N 342 N 403 N 339 ₂₇₇₄	M M Find the time to the Point of Safe Return (PSR). Given: Maximum useable fuel 15000 kg. Minimum reserve fuel 3500 kg. TAS out 425 kt. Head wind component
c d 3 id	375 N 342 N 403 N 339 2774	M M Find the time to the Point of Safe Return (PSR). Given: Maximum useable fuel 15000 kg, Minimum reserve fuel 3500 kg, TAS out 425 kt, Head wind component out 30 kt, TAS return 430 kt, Tailwind component return 20 kt, Average fuel flow
c d 3 id	375 N 342 N 403 N 339 2774	M M Find the time to the Point of Safe Return (PSR). Given: Maximum useable fuel 15000 kg, Minimum reserve fuel 3500 kg, TAS out 425 kt, Head wind component out 30 kt, TAS return 430 kt, Tailwind component return 20 kt, Average fuel flow 2150 kg/h
c d 3 id	375 N 342 N 403 N 339 2774 2 h 59	M M Find the time to the Point of Safe Return (PSR). Given: Maximum useable fuel 15000 kg, Minimum reserve fuel 3500 kg, TAS out 425 kt, Head wind component out 30 kt, TAS return 430 kt, Tailwind component return 20 kt, Average fuel flow 2150 kg/h
c d 3 id a b	375 N 342 N 403 N 339 2774 2 h 59 3 h 43	M Find the time to the Point of Safe Return (PSR). Given: Maximum useable fuel 15000 kg, Minimum reserve fuel 3500 kg, TAS out 425 kt, Head wind component out 30 kt, TAS return 430 kt, Tailwind component return 20 kt, Average fuel flow 2150 kg/h min
c d 3 id a b c	375 N 342 N 403 N 339 2774 2 h 59 3 h 43 2 h 51	M M Find the time to the Point of Safe Return (PSR). Given: Maximum useable fuel 15000 kg, Minimum reserve fuel 3500 kg, TAS out 425 kt, Head wind component out 30 kt, TAS return 430 kt, Tailwind component return 20 kt, Average fuel flow 2150 kg/h min min

3 id	340 4024	Given : Distance A to B 2050 NM Mean groundspeed 'on' 440 kt Mean groundspeed 'back' 540 kt The distance to the point of equal time (PET) between A and B is :
а	920 N	M
b	1025	NM
С	11 30 I	NM
d	1153 N	JM
3 id	341 4025	Given : Distance A to B3060 NM Mean groundspeed 'out'440 kt Meangroundspeed 'back'540 kt Safe Endurance10 hours The time to the Pointof Safe Return (PSR) is:
а	5 hou	s 20 minutes
b	5 hou	s 45 minutes
С	3 hou	rs 55 minutes
d	5 hou	rs 30 minutes
3 id	342 4417	Given : $X = Distance A$ to point of equal time (PET) between A and B E = Endurance D = Distance A to B O = Groundspeed 'on' H = Groundspeed 'back' The formula for calculating the distance X to point of equal time (PET) is:
а	DхН	X = O + H
b	DxO	X = O + H
С	ЕхО	x H X = O + H
d	DxO	x H X = O + H
3 id	343 4418	Given : Course A to B 088° (T) distance 1250 NM Mean TAS 330 kt Mean W/V 340°/60 kt The time from A to the PET between A and B is :
а	1 hou	54 minutes
b	1 hou	r 42 minutes
с	1 hou	39 minutes
d	2 hou	s 02 minutes
3	844	Given Distance X to Y 2700 NM Mach Number 0.75
id	5102	Temperature -45°C Mean wind component 'on' 10 kt tailwind Mean wind compontent 'back' 35 kt tailwind The distance from X to the point of equal time (PET) between X and Y is :
а	1350	JM
b	1386 I	NM
C	1313	JM
d	1425 I	NM .
3 id	345 5520	Given the following: D = flight distance X = distance to Point of Equal Time GSo = groundspeed out GSr = groundspeed return The correct formula to find distance to Point of Equal Time is :
а	X = D	x GSr / (GSo + GSr)
b	X = D	x GSo / (GSo + GSr)

- $c X = (D/2) \times GSo / (GSo + GSr)$
- **d** X = (D/2) + GSr / (GSo + GSr)

33.05.02. Computerised flight planning

33.05.02.01. General principles of present systems

346 Which of the following statements is (are) correct with regard to the advantages of computer flight plans ? 1. The computer can file the ATC flight plan. 2. Wind data used by the computer is always more up-to-date than that available to the pilot.

- a Statement 1 only
- b Statement 2 only
- ${\boldsymbol{c}}\$ Both statements
- d Neither statement

347 Which of the following statements is (are) correct with regard to the operation of flight planning computers ? 1. The computer can file the ATC flight plan. 2. In the event of inflight re-routing the computer produces a new plan.

- a Statement 2 only
- b Statement 1 only
- c Both statements
- d Neither statement

348 Which of the following statements is (are) correct with regard to computer flight ^{id} 5111 plans 1. The computer takes account of bad weather on the route and adds extra fuel. 2. The computer calculates alternate fuel sufficient for a missed approach, climb, cruise, descent and approach and landing at the destination alternate.

- a Statement 1 only
- **b** Both statements
- c Statement 2 only
- d Neither statement

33.06. PRACTICAL COMPLETION OF A "FLIGHT PLAN"

33.06.01. Extraction of data

33.06.01.01. Extraction of navigational data

349 Given: Leg Moulins(N46 24.4 E003 38.0)/Dijon(N47 16.3 E005 05.9). Find: Route designator and total distance

- a UG 21, 26 NM
- **b** D, 44 NM
- c UG 21, 69 NM
- d Direct route, 69 NM

33.06.01.03. Extraction of performance data

- **350** (For this question use Flight Planning Manual MRJT 1 Figure 4.3.6) In order to get alternate fuel and time, the twin -jet aeroplane operations manual graph shall be entered with:
- a Distance (NM), wind component, zero fuel mass.
- **b** Still air distance, wind component, zero fuel mass.
- ${\boldsymbol c}\;$ Flight time, wind component, landing mass at alternate.

d Distance (NM), wind component, landing mass at alternate.

- 351 "Integrated range" curves or tables are presented in the Aeroplane Operations
- id 2836 Manuals. Their purpose is
- **a** to determine the optimum speed considering the fuel cost as well as the time related cost of the aeroplane.
- ${\boldsymbol b}$ to determine the flight time for a certain leg under consideration of temperature deviations.
- c to determine the still air distance for a wind components varying with altitude.
- d to determine the fuel consumption for a certain still air distance considering the decreasing fuel flow with decreasing mass.

352 (For this Question use Fuel Planning MRJT1 Fig. 4.2.2) Find the SHORT

- id 4374 DISTANCE CRUISE ALTITUDE for the twin jet aeroplane. Given: Brake release mass=45000 kg, Temperature=ISA + 20°C, Trip distance=50 Nautical Air Miles (NAM)
- **a** 7500 ft
- b 10000 ft
- **c** 12500 ft
- **d** 11000 ft
- 353 (For this Question use Fuel Planning MRJT1 Fig. 4.2.2) Find the SHORT
 id 4375 DISTANCE CRUISE ALTITUDE for the twin jet aeroplane. Given: Brake release mass=40000 kg, Temperature=ISA + 20°C, Trip distance=150 Nautical Air Miles (NAM)
- a 30000 ft
- **b** 25000 ft
- **c** 21000 ft
- **d** 27500 ft

354 (For this Question use Fuel Planning MRJT1) Given: twin jet aeroplane, Zero fuel mass 50000 kg, Landing mass at alternate 52000 kg, Final reserve fuel 2000 kg, Alternate fuel 1000 kg, Flight to destination: Distance 720 NM, True course 030°, W/V 340°/30 kt, Long range cruise, FL 330, Outside air temperature -30°C Find: Estimated trip fuel and time with simplified flight planning

- a 4750 kg, 02 h 00 min
- **b** 4400 kg, 02 h 05 min
- c 4800 kg, 01 h 51 min
- d 4600 kg, 02 h 05 min

355 (For this Question use Fuel Planning MRJT1) Given: Twin jet aeroplane, Ground distance to destination aerodrome is 1600 NM, Headwind component 50 kt, FL 330, Cruise .78 Mach, ISA Deviation +20°C and Landing mass 55000 kg Find: Fuel required and trip time with simplified flight planning

- a 11400 kg, 04 h 12 min
- b 12400 kg, 04 h 00 min
- c 11600 kg, 04 h 15 min
- **d** 12000 kg, 03 h 51 min

 356 (For this Question use Fuel Planning MRJT1) Given: twin jet aeroplane, Dry
 id 4556 operating mass 35500 kg, Traffic load 14500 kg, Final reserve fuel 1200 kg, Distance to alternate 95 NM, Tailwind component 10 kt Find: Fuel required and trip time to alternate with simplified flight planning (ALTERNATE PLANNING)

- a 800 kg, 0.4 hr
- **b** 1000 kg, 40 min
- c 800 kg, 24 min
- d 1000 kg, 24 min

357 (For this Question use Fuel Planning MRJT1) Given: twin jet aeroplane, Estimated ^{id} ⁴⁵⁵⁷ mass on arrival at the alternate 50000 kg, Estimated mass on arrival at the destination 52525 kg, Alternate elevation MSL, Destination elevation 1500 ft Find: Final reserve fuel and corresponding time

- a 2360 kg, 01 h 00 min
- **b** 2360 kg, 30 min
- c 1180 kg, 30 min
- **d** 1180 kg, 45 min

358 (For this Question use Fuel Planning MRJT1) Given: twin jet aeroplane, Estimated ¹⁴⁵⁵⁸ mass on arrival at the alternate 50000 kg, Elevation at destination aerodrome 3500 ft, Elevation at alternate aerodrome 30 ft Find: Final reserve fuel

- **a** 1150 kg
- **b** 2360 kg
- c 1180 kg
- d 2300 kg

33.06.01.05. Completion of fuel plan

3 id	359 1859	Given: Dry operating mass (DOM)= 33500 kg Load= 7600 kg Maximum allowable take-off mass= 66200 kg Standard taxi fuel= 200 kg Tank capacity= 16 100 kg The
		maximum possible take-off fuel is:
а	15 90	0 kg
b	17 10	0 kg
С	16 30	0 kg
d	17 30	0 kg
3 id	2044	(For this question Flight Planning Manual MRJT 1 Figure 4.5.3.1) Given: long range cruise: temp63° C at FL 330 initial gross mass enroute 54 100 kg; leg flight time 29 min Find: fuel consumption for this leg
а	1 200	kg
b	1 100	kg
С	1 020	kg
d	1 680	kg
id	2045	(For this question Flight Planning Manual MRJT 1 Figure 4.5.3.1) Given: flight time from top of climb to the enroute point in FL280 is 48 min. Cruise procedure is long range cruise (LRC). Temp. ISA -5° C Take-off mass 56 000 kg Climb fuel 1 100 kg Find: distance in nautical air miles (NAM) for this leg and fuel consumption:
а	345 N	AM; 2000 kg
b	350 N	AM; 2000 kg
С	345 N	AM; 2100 kg
d	437 N	AM; 2100 kg
3 id	2046	(For this question use Flight Planning Manual MRJT 1 Figure 4.5.1) Given: estimated take-off mass 57 500 kg; initial cruise FL 280; average temperature during climb ISA -10°C; average head wind component 18 kt Find: climb time
а	15 mir	1
b	11 mir	1
С	13 mi	n
d	14 mir	ו
3 id	363 2047	(For this question use Flight Planning Manual MRJT 1 Figure 4.5.1) Given : brake release mass 57 500 kg temperature ISA -10°C; head wind component 16 kt initial FL 280 Find: still air distance (NAM) and ground distance (NM) for the climb
а	59 NA	M;62 NM
b	62 NA	M; 59 NM
С	67 NA	M; 71 NM
d	71 NA	M;67 NM
3 id	2048	(For this question use Flight Planning Manual MRJT 1 Figure 4.5.1) Given : mass at brake release 57 500 kg; temperature ISA -10°C; average head wind component 16 kt initial cruise FL 280 Find: climb fuel
а	1138	kg
b	1238	kg
С	1387	kg
d	1040	kg

365	(For this question Flight Planning Manual MRJT 1 Figure 4.5.3.1) Given :FL 330;
id 2050	long range cruise; OAT -63°C; gross mass 50 500 kg. Find: true airspeed (TAS)

- a 420 kt
- **b** 433 kt
- **c** 431 kt
- **d** 418 kt

366 (For this question Flight Planning Manual MRJT 1 Figure 4.5.3.1) Given: long range cruise; outside air temperature (OAT) -45 ° C in FL 350; mass at the beginning of the leg 40 000 kg; mass at the end of the leg 39 000 kg. Find: true airspeed (TAS) at the end of the leg and the distance (NAM).

- a TAS 423 kt; 936 NAM
- **b** TAS 423 kt; 227 NAM
- c TAS 431 kt; 1163 NAM

d TAS 431 kt; 227 NAM

367 (For this question use Flight Planning Manual MRJT 1) Given: estimated take-off mass 57 000 kg; still air distance 150 NAM; outside air temperature (OAT) ISA -

- 10K; cruise at 0.74 Mach. Find : cruise altitude and expected true airspeed
- a 22 000 ft; 451 kt
- **b** 25 000 ft; 445 kt
- **c** 22 000 ft; 441 kt
- d 25 000 ft; 435 kt

368 (For this question use Flight Planning Manual MRJT 1 Figure 4.4) Planning a flight

id 2331 from Paris Charles de Gaulle to London Heathrow for a twin - jet aeroplane.
 Preplanning: Dry Operating Mass (DOM): 34 000 kg Traffic Load: 13 000 kg The holding is planned at 1 500 ft above alternate elevation. The alternate elevation is 256 ft. The holding is planned for 30 minutes with no

- a 48 675 kg.
- **b** 49 250 kg.
- c 2 250 kg.

d 48 125 kg.

369 (For this question use Route Manual chart E(HI)4) Planning a flight from Paris
 ²³³² Charles de Gaulle (N49 00.9 E002 36.9) to London Heathrow (N51 29.2 W000 27.9) for a twin - jet aeroplane. The alternate airport is Manchester (N53 21.4 W002 15.7) Preplanning: The wind from London to Manchester is 250°/30 kt The distance from London to Manchester is 160 NM. Assume the Estimat

- a 1 200 kg and 26 minutes.
- b 1 300 kg and 28 minutes.
- c 1 600 kg and 36 minutes.
- d 1 450 kg and 32 minutes.

370 (For this question use Route Manual chart E(HI)4) Planning a flight from Paris

- ^{id} ²³³³ Charles de Gaulle (N49 00.9 E002 36.9) to London Heathrow (N51 29.2 W000 27.9) for a twin jet aeroplane. Preplanning: Powersetting: Mach= 0.74 Planned flight level FL 280 The Landing Mass in the fuel graph is 50 000 kg The trip distance used for calculation is 200 NM The wind from P
- a 1 450 kg.
- **b** 1 550 kg.
- **c** 1 900 kg.
- d 1740 kg.
- 371 (For this Question use Fuel Planning MRJT1) Finish the ENDURANCE/FUEL
 ^{id} 2767 CALCULATION and determine ATC ENDURANCE for a twin jet aeroplane, with the help of the table provided. Contingency is 5% of the planned trip fuel and fuel flow for extra fuel is 2400 kg/h.
- a ATC ENDURANCE: 04:07
- **b** ATC ENDURANCE: 03:52
- c ATC ENDURANCE: 03:37
- d ATC ENDURANCE: 04:12
- **372** (For this Question use Fuel Planning MRJT1) The aeroplane gross mass at top of climb is 61500 kg. The distance to be flown is 385 NM at FL 350 and OAT -54.3 °C. The wind component is 40 kt tailwind.Using long range cruise procedure what fuel is required?
- a 2350 kg
- **b** 2250 kg
- c 2150 kg
- **d** 2050 kg
- 373 (For this Question use Fuel Planning MRJT1) Find: Time, Fuel, Still Air Distance
- id 2773 and TAS for an enroute climb 280/.74 to FL 350. Given: Brake release mass 64000 kg, ISA +10°C, airport elevation 3000 ft
- a 25 min, 1875 kg, 148 Nautical Air Miles (NAM), 391 kt
- b 26 min, 2050 kg, 157 Nautical Air Miles (NAM), 399 kt
- c 20 min, 1750 kg, 117 Nautical Air Miles (NAM), 288 kt

d 26 min, 1975 kg, 157 Nautical Air Miles (NAM), 399 kt

- **374** For flight planning purposes the landing mass at alternate is taken as:
- **id** 4370
- a Landing Mass at destination plus Alternate Fuel.

b Zero Fuel Mass plus Final Reserve Fuel.

- c Zero Fuel Mass plus Final Reserve Fuel and Alternate Fuel.
- d Zero Fuel Mass plus Final Reserve Fuel and Contingency Fuel.
- **375** Given: Maximum allowable take-off mass 64400 kg, Maximum landing mass 56200
- id 4552 kg, Maximum zero fuel mass 53000 kg, Dry operating mass 35500 kg, Traffic load 14500 kg, Trip fuel 4900 kg, Minimum Take-off Fuel 7400 kg Find: Maximum allowable take-off fuel
- **a** 11400 kg
- b 11100 kg
- **c** 14400 kg
- **d** 8600 kg

3	76	Given: Maximum allowable take-off mass 64400 kg, Maximum landing mass 56200
id	4553	kg, Maximum zero fuel mass 53000 kg, Dry operating mass 35500 kg, Traffic load
		14500 kg, Trip fuel 4900 kg, Take-off fuel 7400 kg Find: Maximum additional load
а	5600	kg
b	4000	kg
С	7000	kg
d	3000	kg
3	77	(For this Question use Fuel Planning MRJT1) Planning an IFR-flight from Paris to
id	4559	London for a twin jet aeroplane. Given: Estimated Take-off Mass (TOM) 52000 kg,
		Airport elevation 387 ft, FL 280, W/V 280°/40 kt, ISA-Deviation -10°C, Average True
_		Course 340 Find. Ground distance to the top of climb (TOC)
а	53 NI	1
b	56 NI	1
С	50 NN	1
d	47 NN	
3	78	(For this Question use Fuel Planning MRJT1) Planning an IFR-flight from Paris
iu	4500	Estimated Take-off Mass (TOM) 52000 kg. Airport elevation 387 ft. FL 280. W/V
		280°/40 kt, ISA Deviation -10°C, Average True Course 340° Find: Time to the top of
		climb (TOC)
а	12 mi	n
b	3 min	
С	11 mi	n
d	15 mi	n
3	79	(For this Question use Fuel Planning MRJT1) Planning an IFR-flight from Paris to
id	4561	London for the twin jet aeroplane. Given: Estimated Take-off Mass (TOM) 52000
		kg, Airport elevation 387 ft, FL 280, W/V 280°/40 kt, ISA Deviation -10°C, Average
~	1100	
a h	1000	
с С	1000	
ט ה	1500	
	1000	US
ت id	4568	London for the twin jet aeroplane. Given: Gross mass 50000 kg. FL 280, ISA
		Deviation -10°C, Cruise procedure Mach 0.74 Determine the TAS
а	417 k	t
b	440 k	t
с	427 k	t
d	430 k	t
.3	81	When calculating the fuel required to carry out a given flight. one must take into
id	5523	account : 1 - the wind 2 - foreseeable airborne delays 3 - other weather forecasts
		4 - any foreseeable conditions which may delay landing The combination which
		provides the correct statement is :
а	1 - 2	- 3
b	1 - 3	
С	2 - 4	
d	1 - 2	- 3 - 4

33.06.01.06. Computation of CP (critical point)

382 Find the distance to the POINT OF SAFE RETURN (PSR). Given: maximum useable fuel 15000 kg, minimum reserve fuel 3500 kg, Outbound: TAS 425 kt, head wind component 30 kt, fuel flow 2150 kg/h, Return: TAS 430 kt, tailwind component 20 kt, fuel flow 2150 kg/h

- a 1463 NM
- **b** 1143 NM
- c 1125 NM
- **d** 1491 NM

33.06.01.07. Completion of air traffic flight plan

383 On an ATC flight plan, an aircraft indicated as "H" for "Heavy"

id 5513

- a has a certified take-off mass greater than or equal to 140 000 kg
- b has a certified landing mass greater than or equal to 136 000 kg
- c is of the highest wake turbulence category
- d requires a runway length of at least 2 000m at maximum certified take-off mass

384 On a VFR flight plan, the total estimated time is:

id 5516

a the estimated time from take-off to overhead the destination airport

- **b** the estimated time from take-off to overhead the destination airport, plus 15 minutes
- ${\boldsymbol{c}}\,$ the estimated time from take-off to landing at the alternate airport
- ${\bf d}\,$ the estimated time from engine start to landing at the destination airport

385 On an ATC flight plan, the letter "Y" is used to indicate that the flight is carried out under the following flight rules.

a IFR followed by VFR

- **b** VFR followed by IFR
- c IFR
- $\boldsymbol{d} \ \ \mathsf{VFR}$

386 On an ATC flight plan, to indicate that you will overfly the way-point TANGO at 350 kts at flight level 280, you write:

- a TANGO / K0350 FL280
- b TANGO / N0350 F280
- c TANGO / FL280 N0350
- d TANGO / KT350 F280

387 On a ATC flight plan, to indicate that you will overfly the way-point ROMEO at 120 kt at flight level 085, you will write :

- a ROMEO / K0120 FL085
- b ROMEO / N0120 F085
- **c** ROMEO / FL085 N0120
- d ROMEO / F085 N0120